



Design of Airfoils for Wind Turbine Blades

Ruud van Rooij
(r.vanrooij@citg.tudelft.nl)
Nando Timmer

Delft University of Technology
The Netherlands



Delft University of Technology

13200 Bsc+ Msc students, 4750 employees

DUWIND

Delft University Wind Energy Research
Institute

(Coordinator: Section Wind Energy)

Faculties:

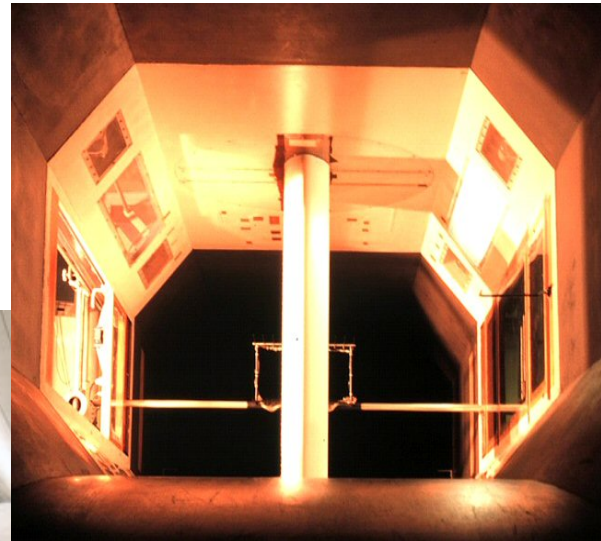
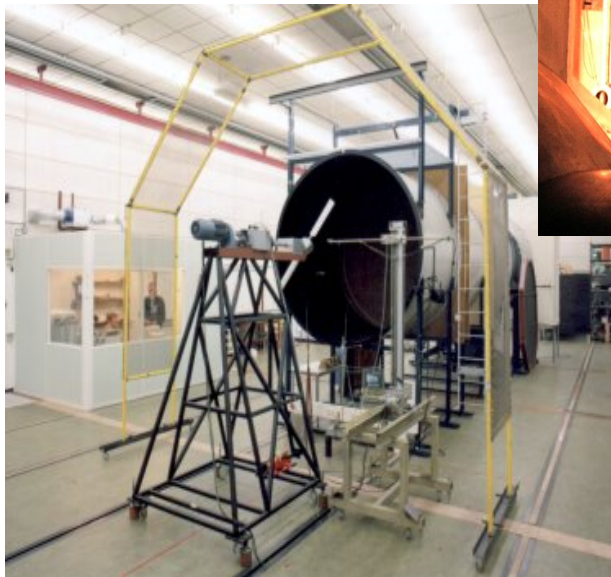
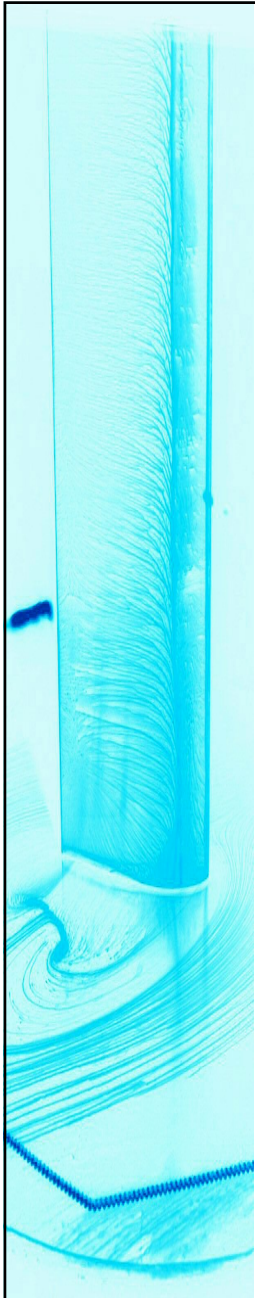
- **Civil Engineering and Geosciences** (Wind Energy, Offshore)
<http://www.windenergy.citg.tudelft.nl/home/flash/index.html>
- **Information Technology and Systems** (Electrical group)
- **Design, Engineering and Production** (Systems & Control)
- **Aerospace Engineering** (Aero, Aeroelastics)

Section Wind Energy

(Civil Engineering and Geosciences => Aerospace Engineering)

Aerodynamic research

- Facilities



low speed
wind-tunnel

open-jet
wind tunnel



research wind turbine

Contents

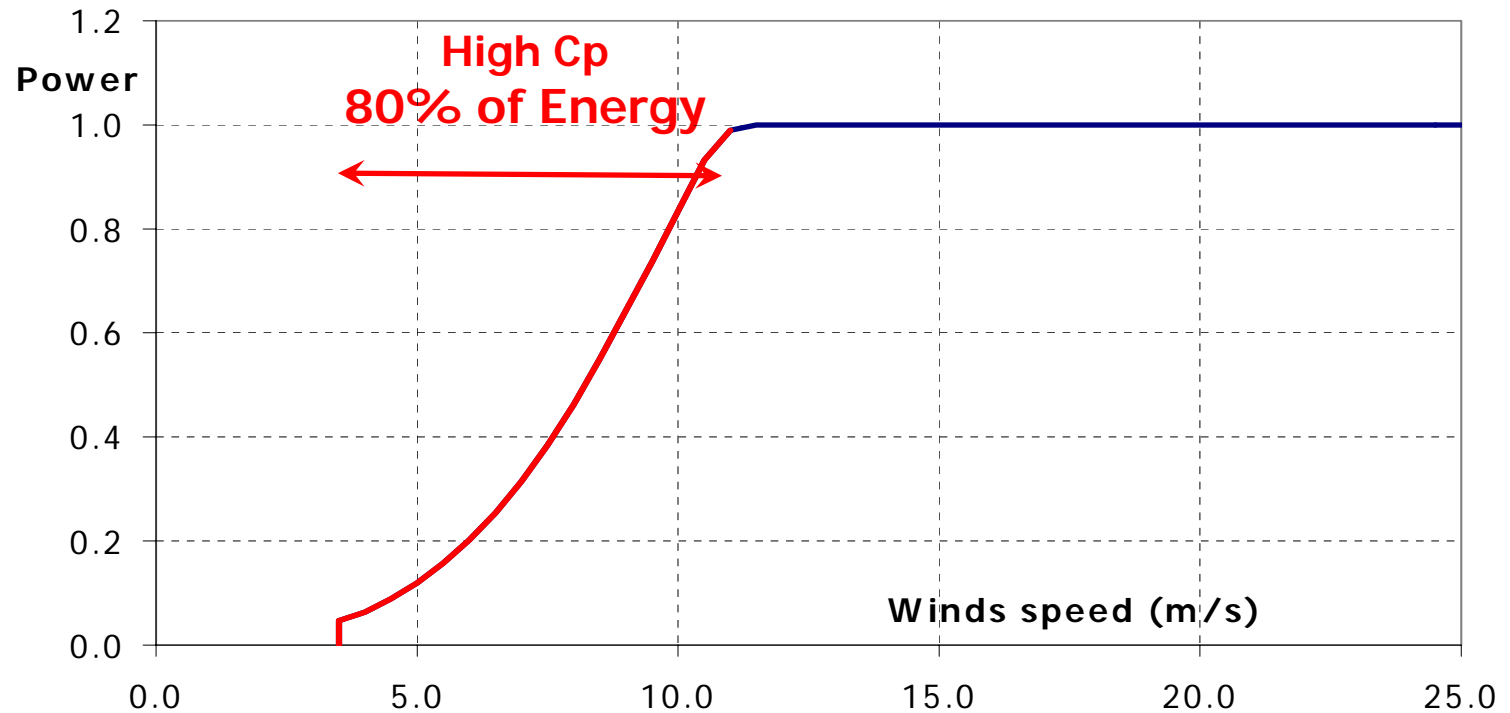
- **Background**
- **Design goals HAWT airfoils**
- **Design approach**
 - Performance comparison
- **Airfoil testing**
- **Effect on wind turbine power C_p**
- **Overview HAWT airfoils**

Background Operational area

Control:

Variable RPM

Power restriction

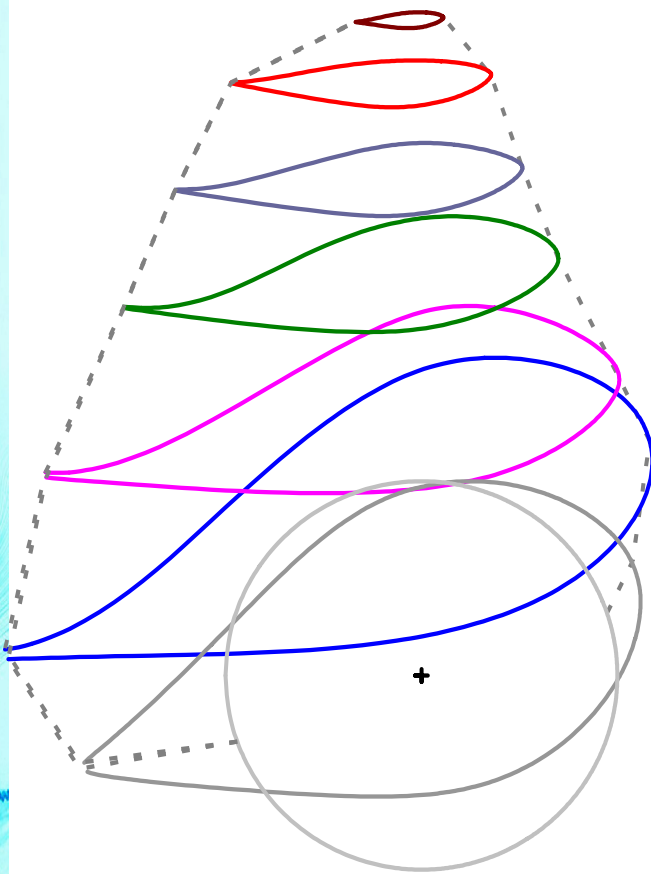
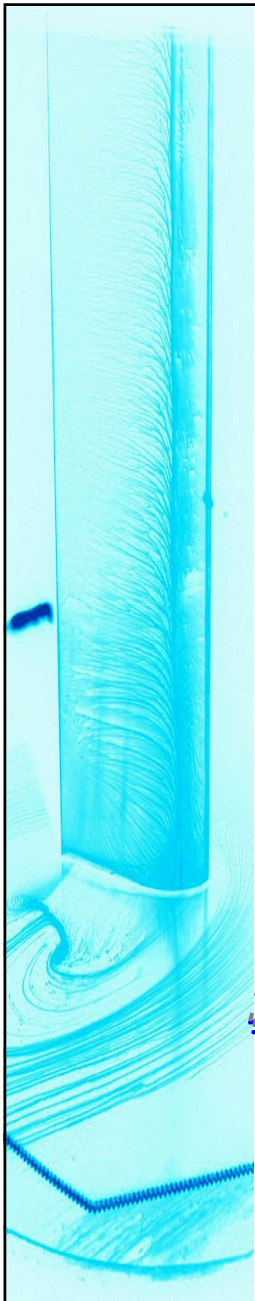


Airfoil:

High max. L/D

Max. lift considerations

Background Blade geometry



Structural:

Outboard: $t/= .15-18$

Mid span: $t/= .25$

Inboard: $t/> .30$

Transition piece

Airfoil:

- High max. L/D
- Insensitive to roughness
- Similar design angle
- High max. lift (Rot. Effects)

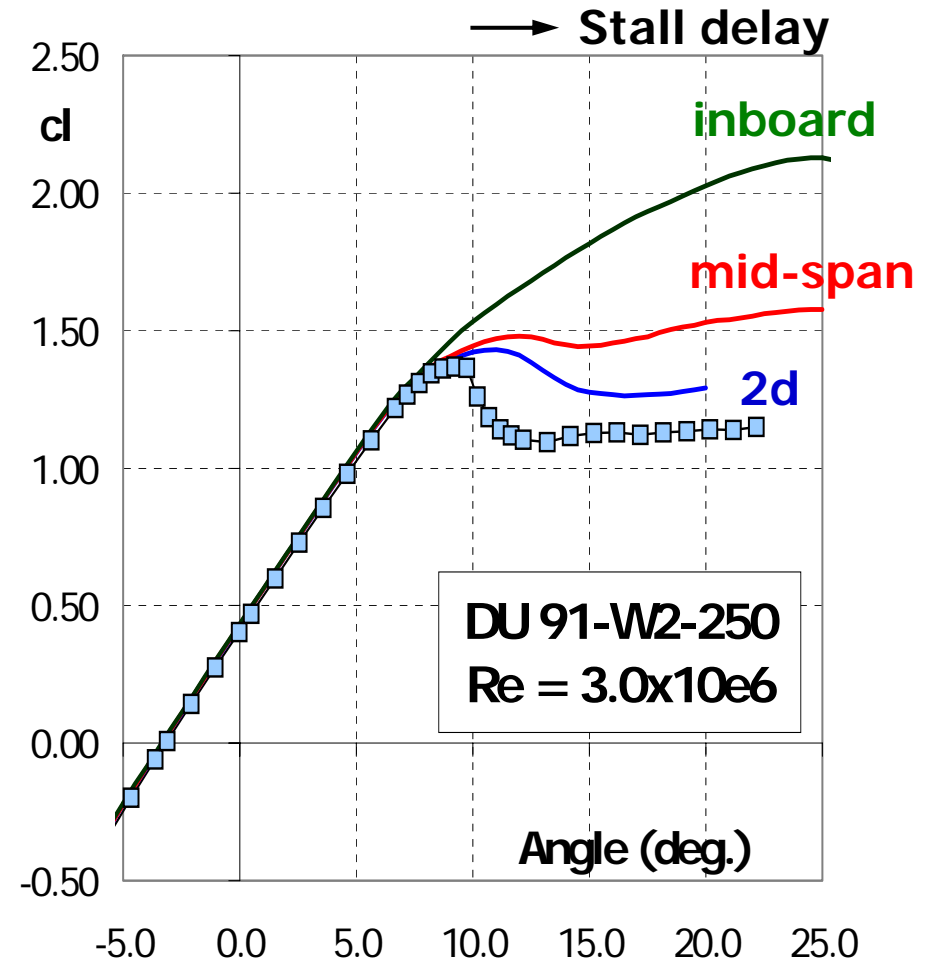
No Aerodynamic demands

Background Effect of rotation

RFOIL code

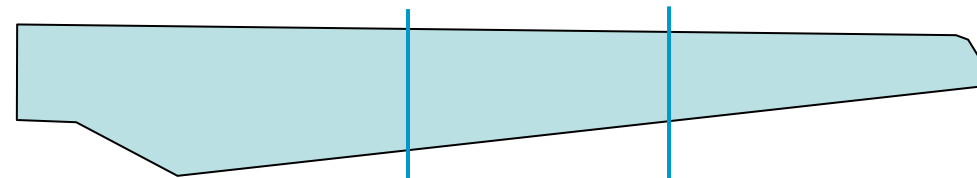
- Integral boundary layer eq.
- Extended for radial flow
 - Radial equations
 - Cross flow profile

parameter is c/r
(= local solidity)



Design goals HAWT airfoils

steady

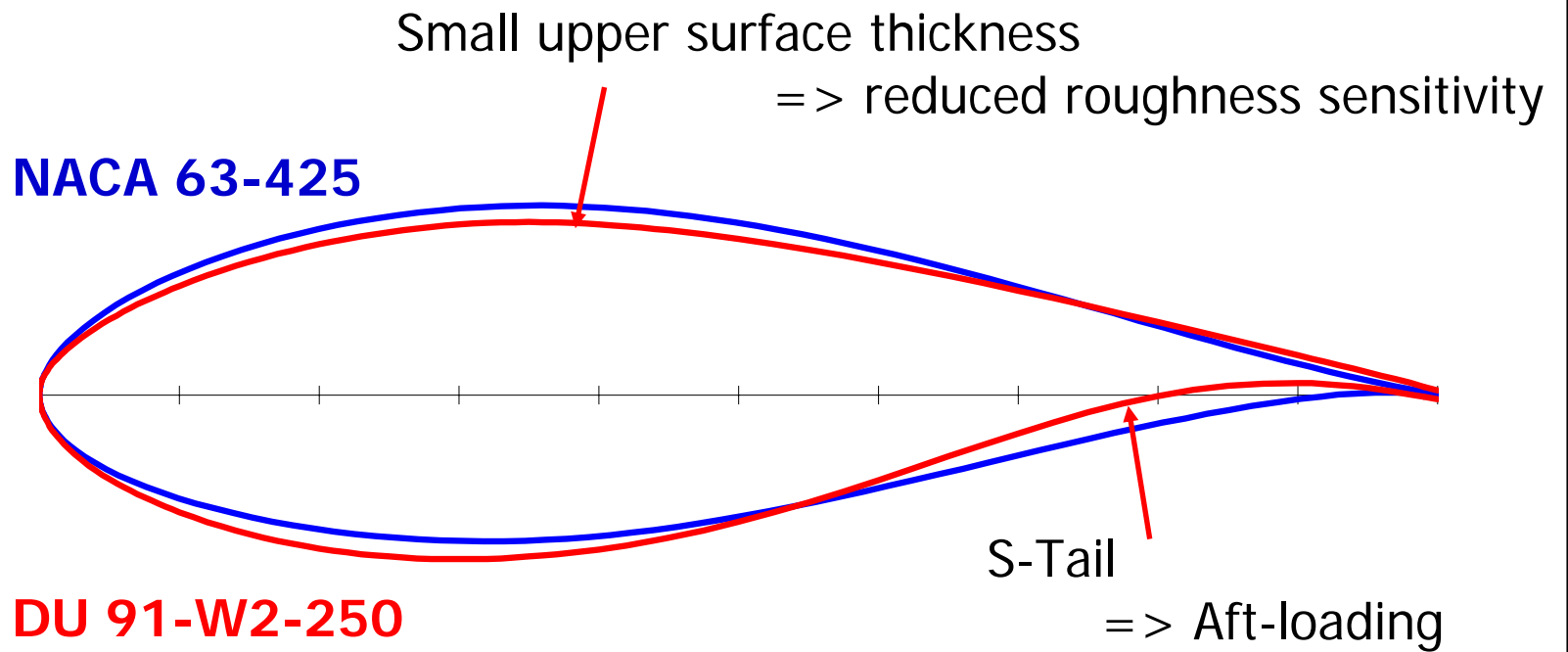


	> .28	.28 - .21	.21 >
Thickness-to-chord ratio	> .28	.28 - .21	.21 >
High maximum lift-to-drag ratio	●	● ●	● ● ●
Low max. and benign post stall			● ●
Insensitivity to roughness	●	● ●	● ● ●
Low noise		●	● ● ●
Geometric compatibility	● ●	● ●	● ●
Structural demands	● ● ●	● ●	●

Design approach

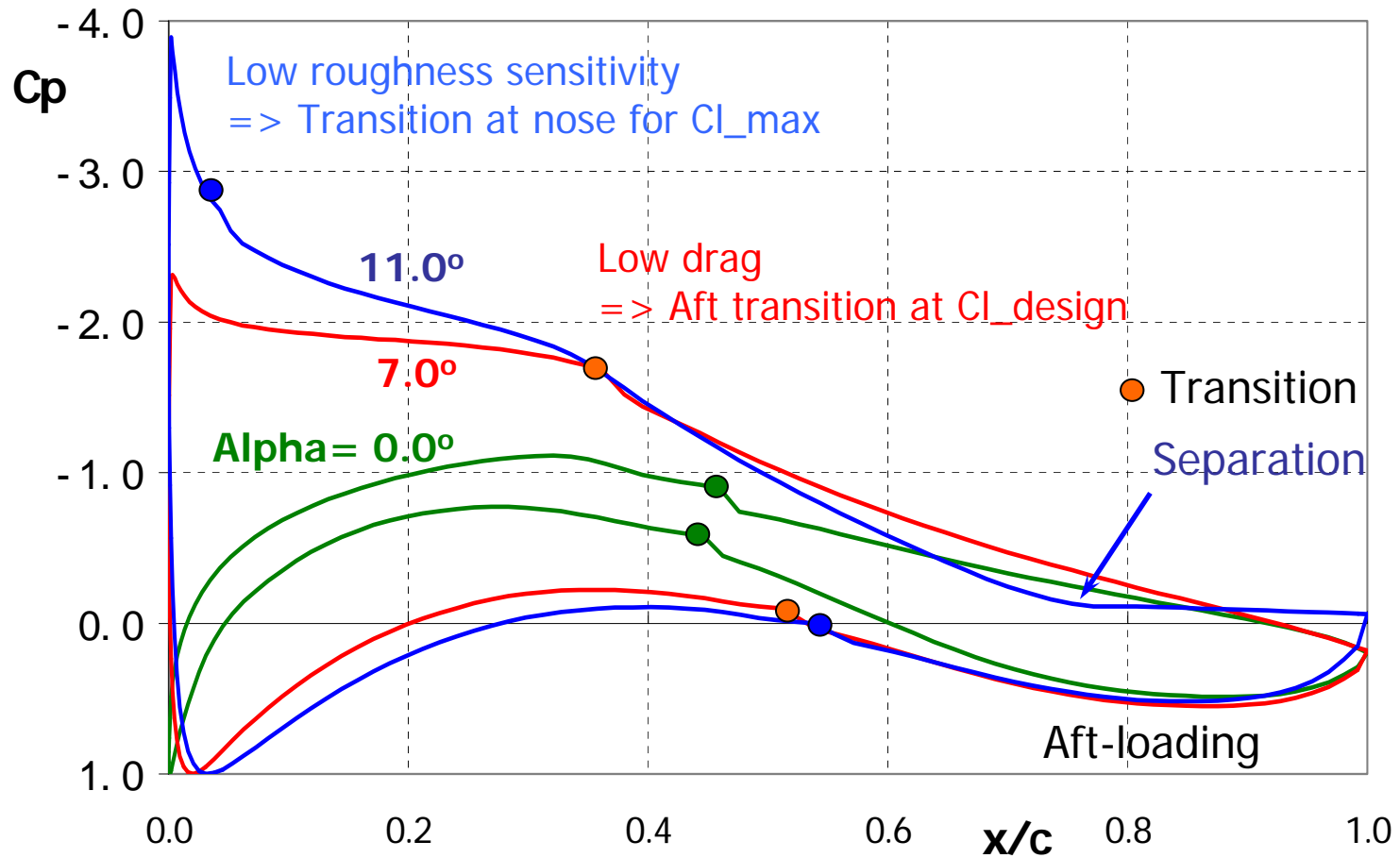
(example DU 91-W2-250)

Main features



Design approach

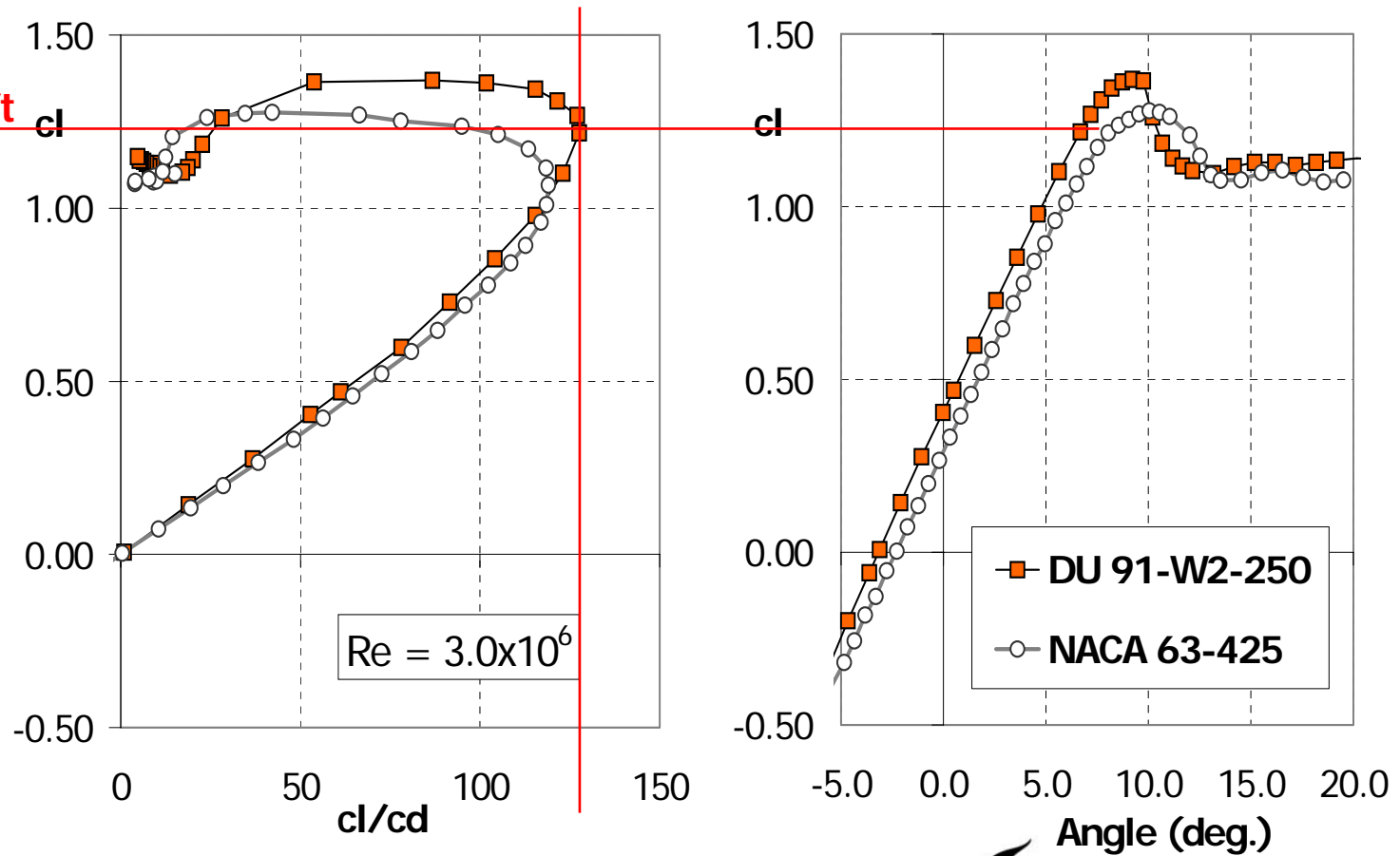
(pressure distributions DU 91-W2-250, $Re = 3.0 \times 10^6$)



Airfoil design (2d performance)

Measurements at LST-TU Delft: Clean

Design lift

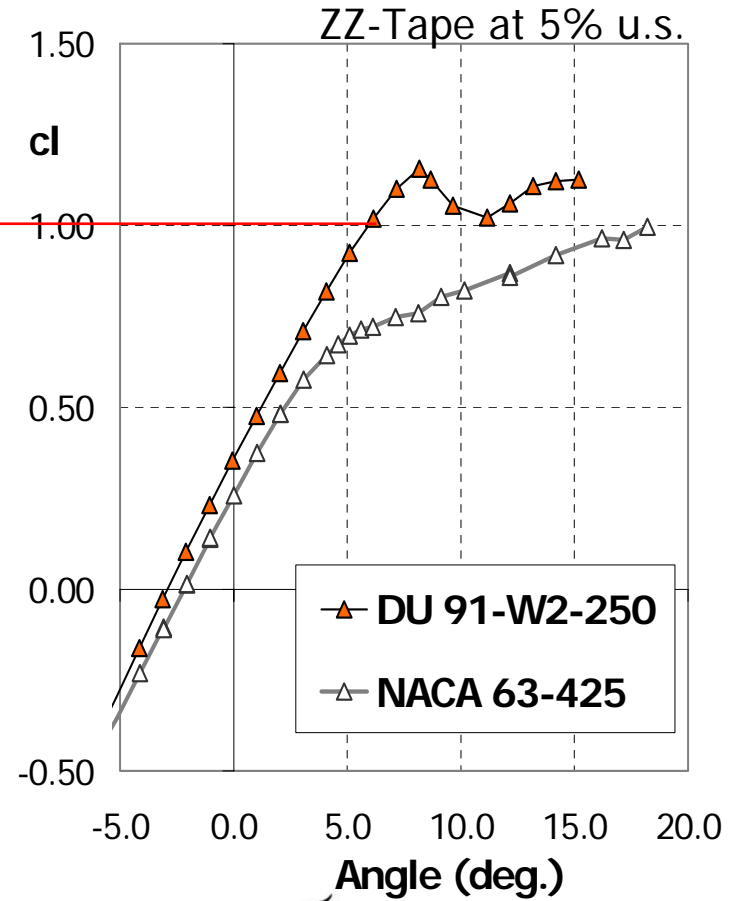
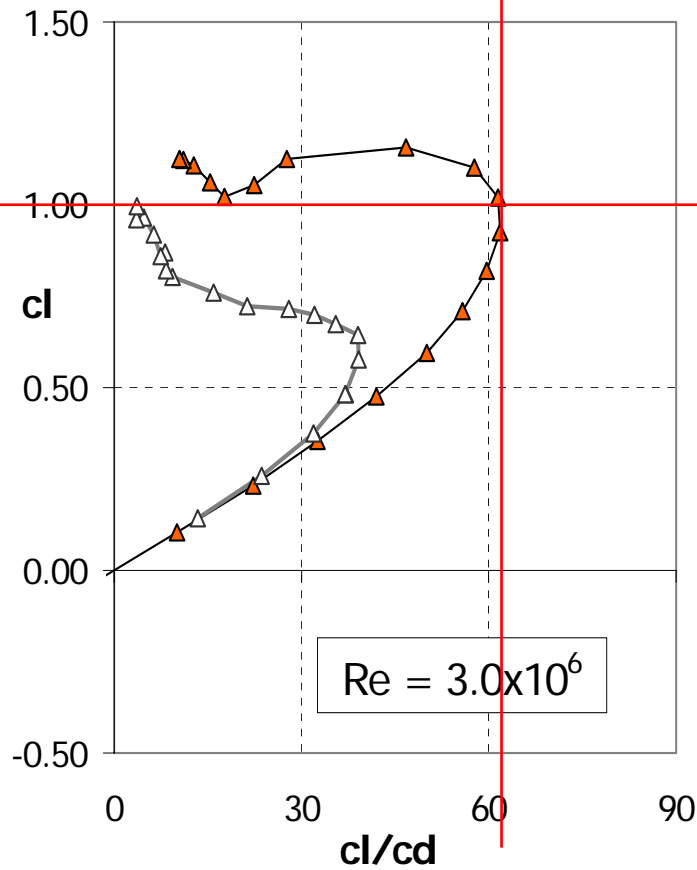


Airfoil design

(2d performance)

Measurements at LST-TU Delft: Roughness simulated

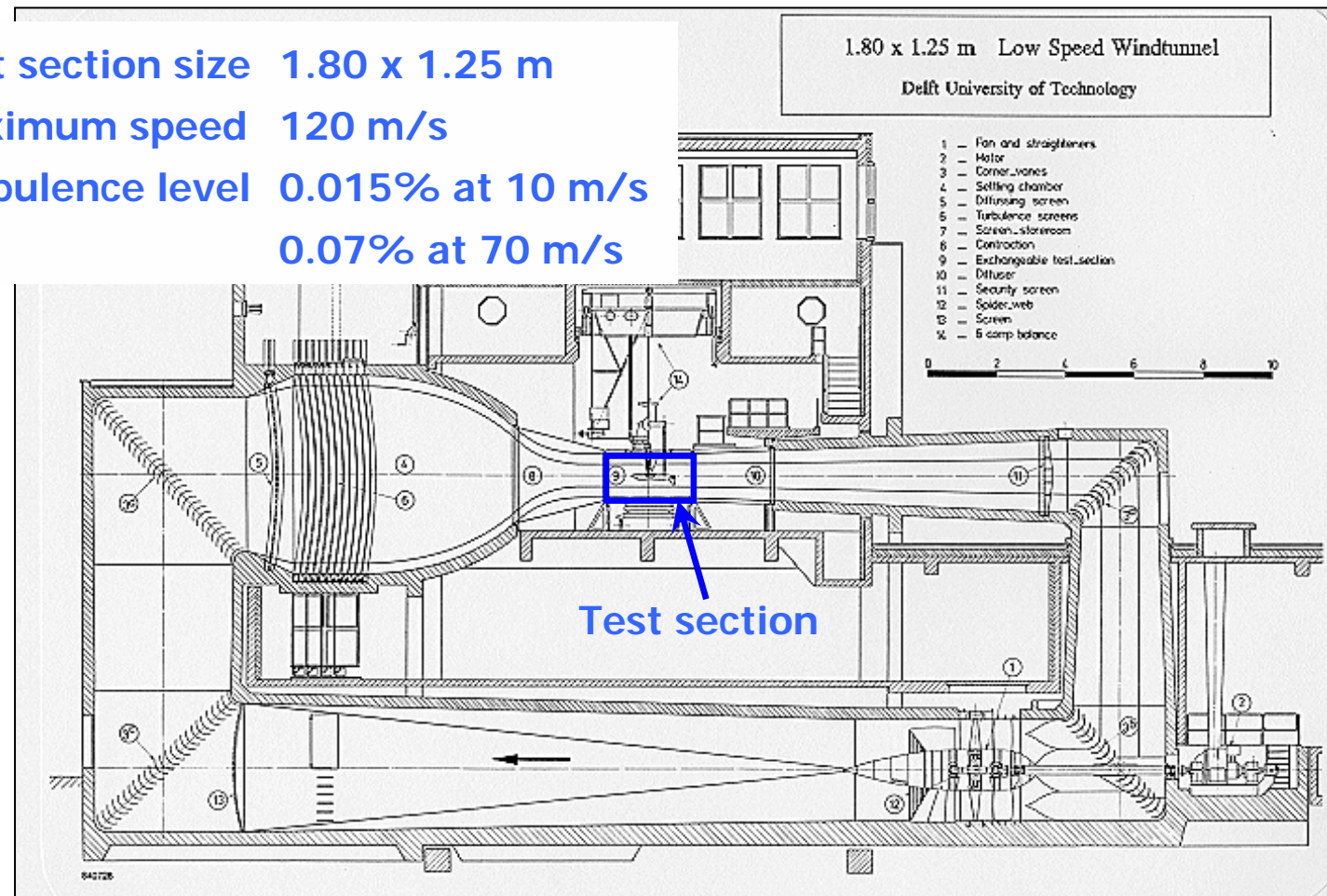
Design lift



Airfoil testing

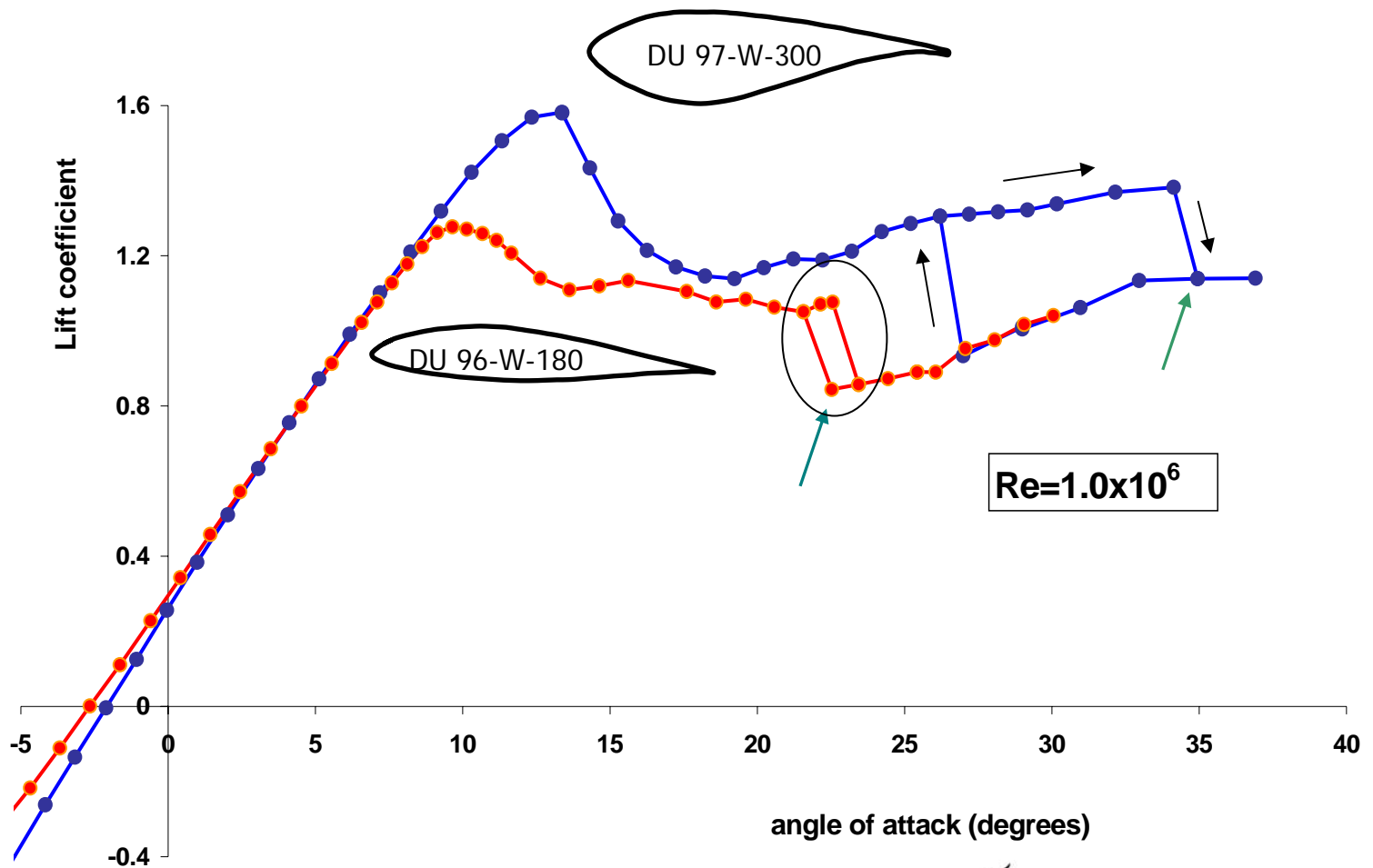
(Low speed low turbulence tunnel)

Test section size 1.80 x 1.25 m
Maximum speed 120 m/s
Turbulence level 0.015% at 10 m/s
0.07% at 70 m/s



Airfoil testing

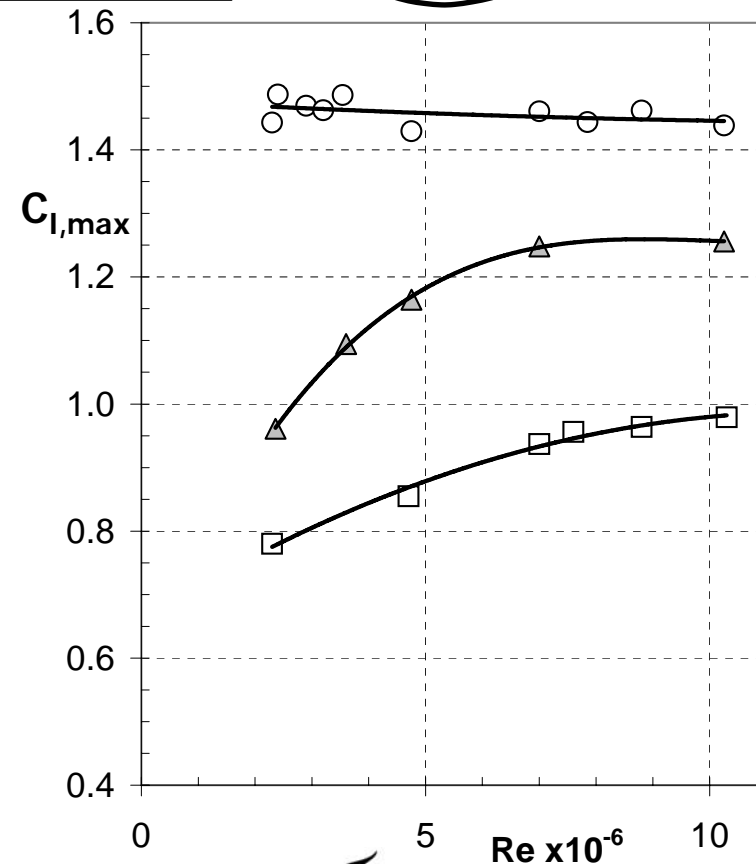
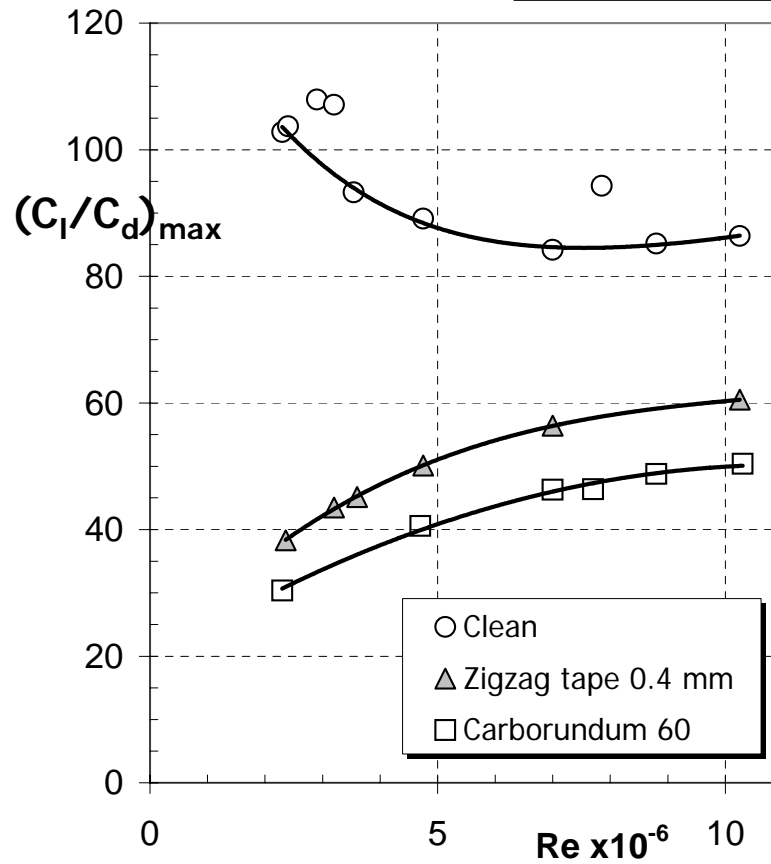
(effect of leading edge thickness)



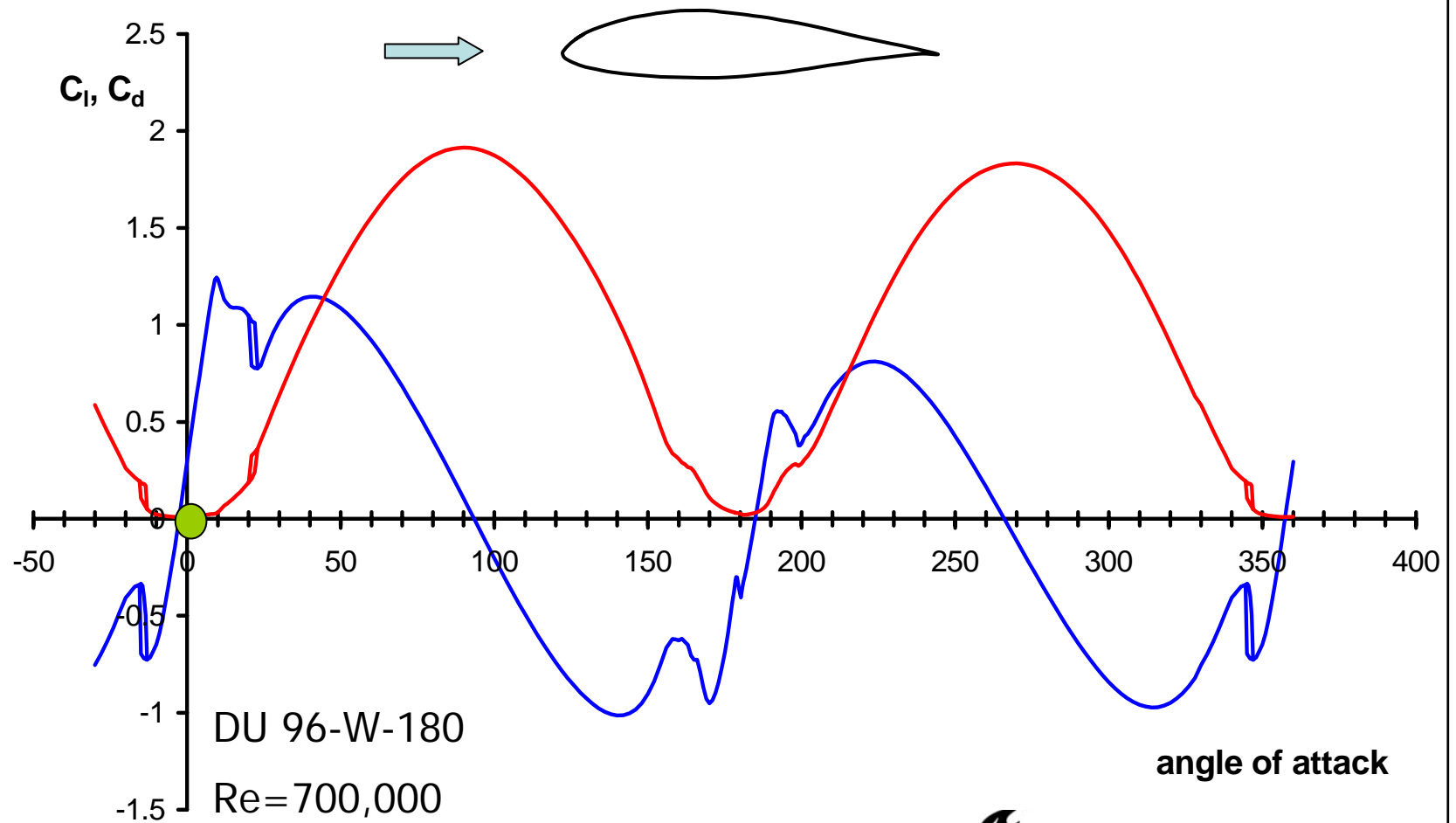
Airfoil testing

(effect of high Reynolds numbers)

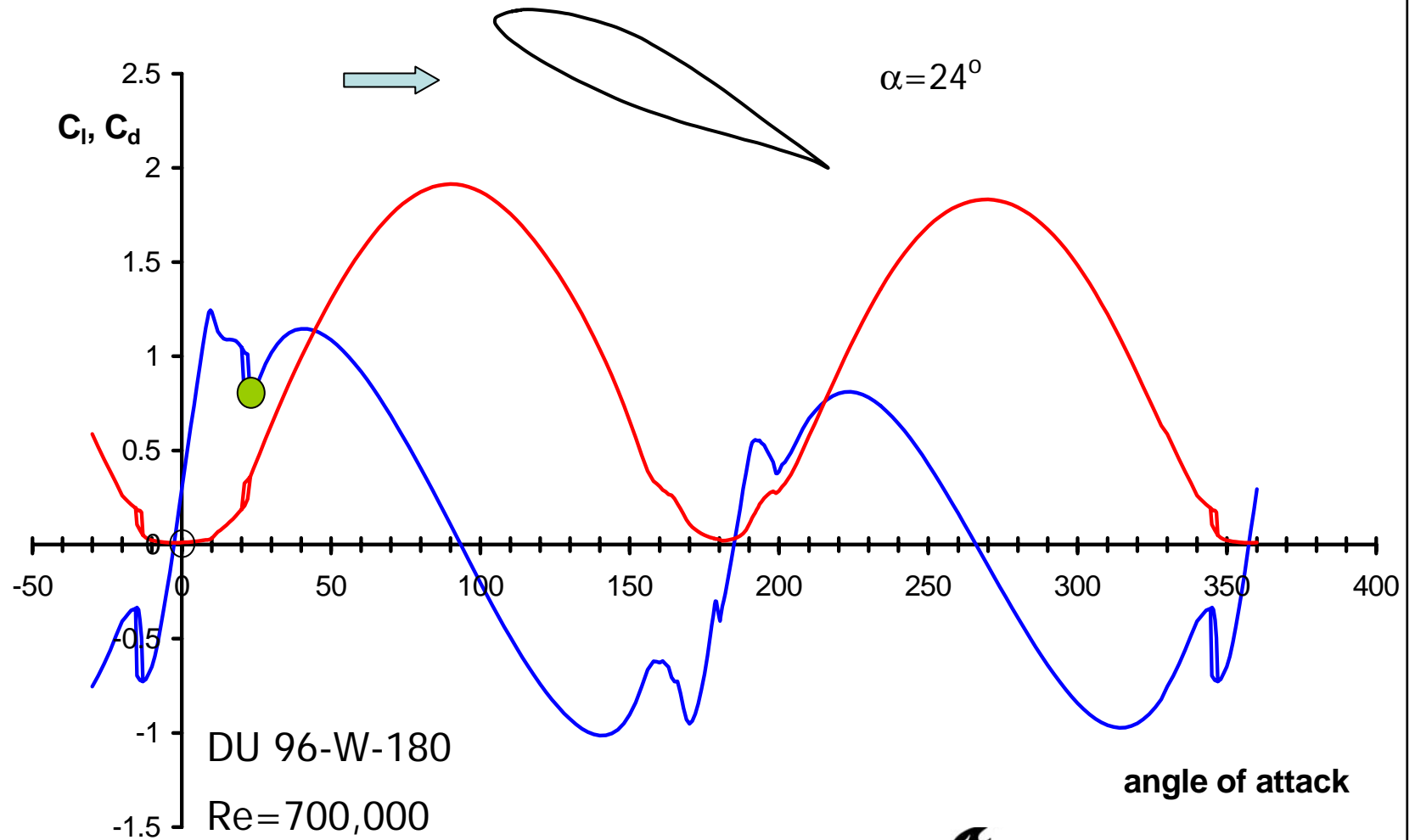
Airfoil: DU 97-W-300Mod



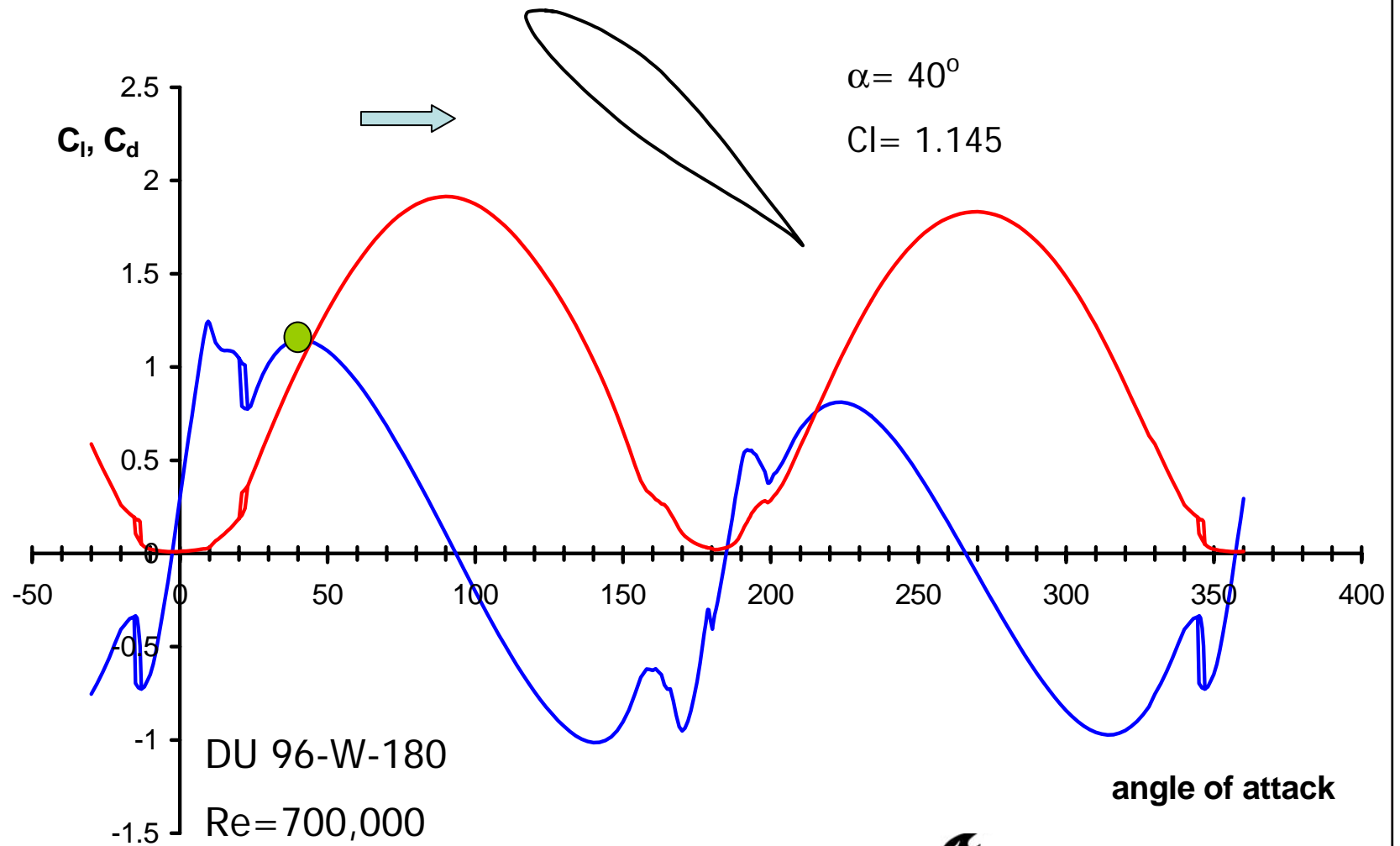
Airfoil testing (360 degrees)



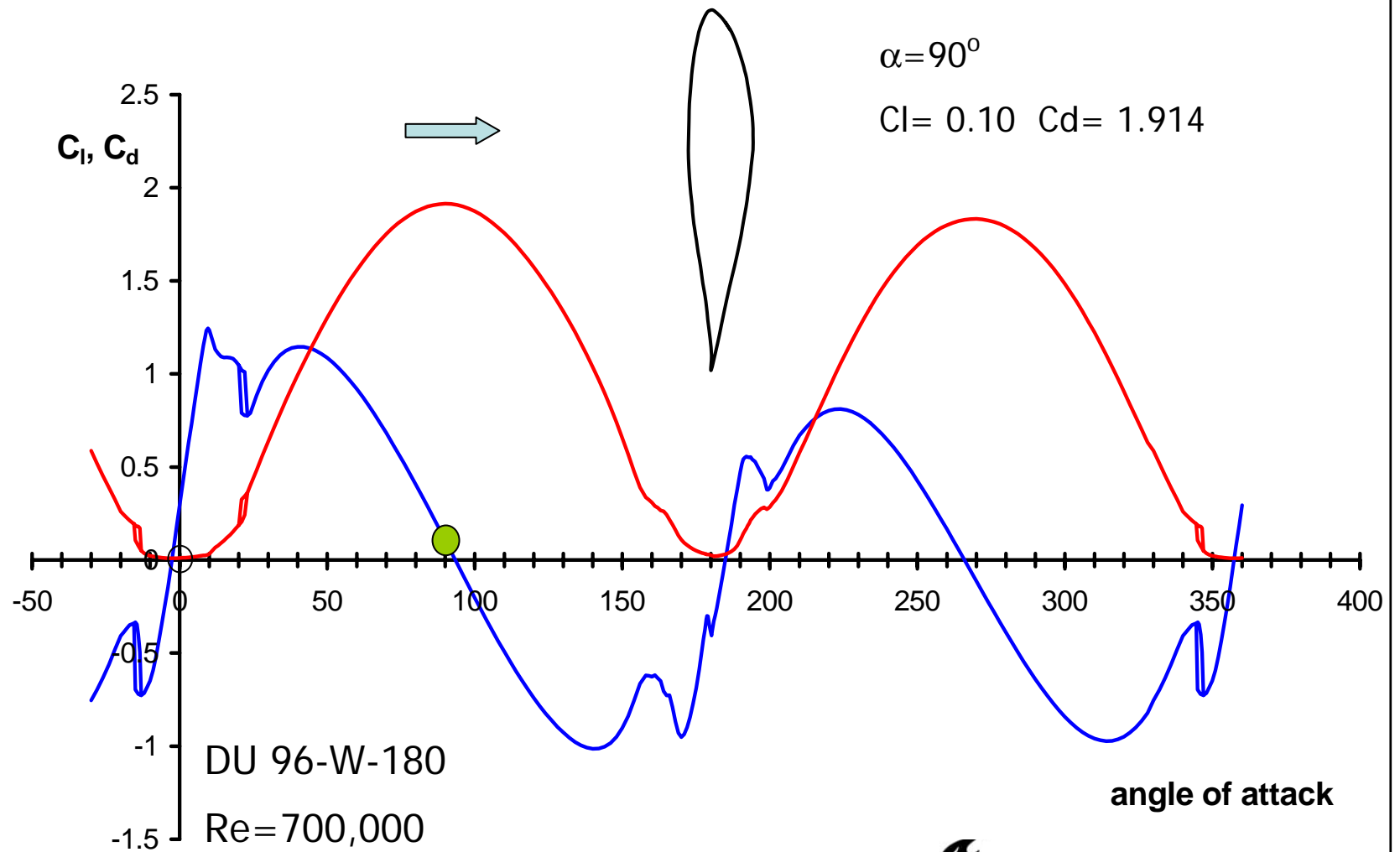
Airfoil testing (360 degrees)



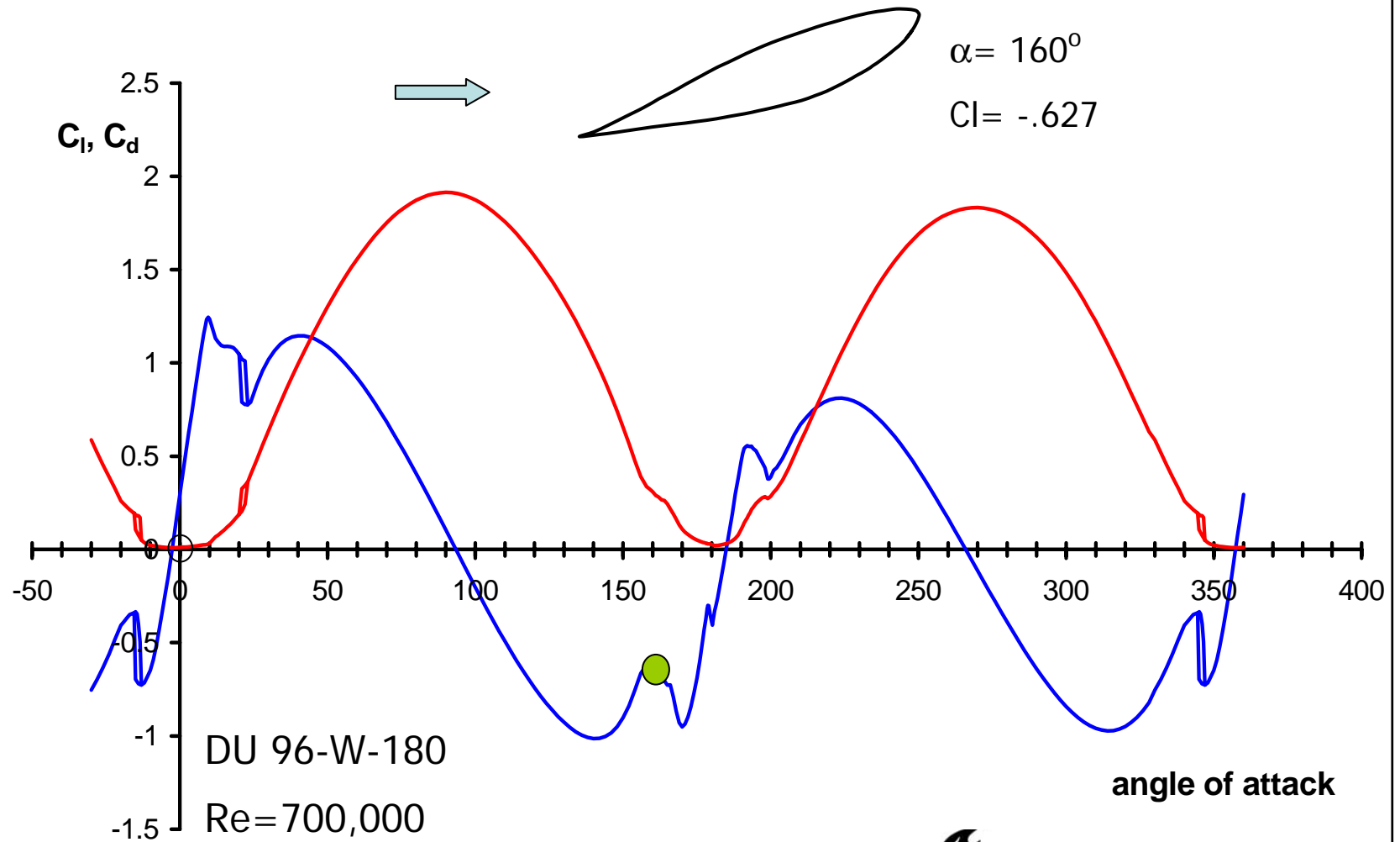
Airfoil testing (360 degrees)



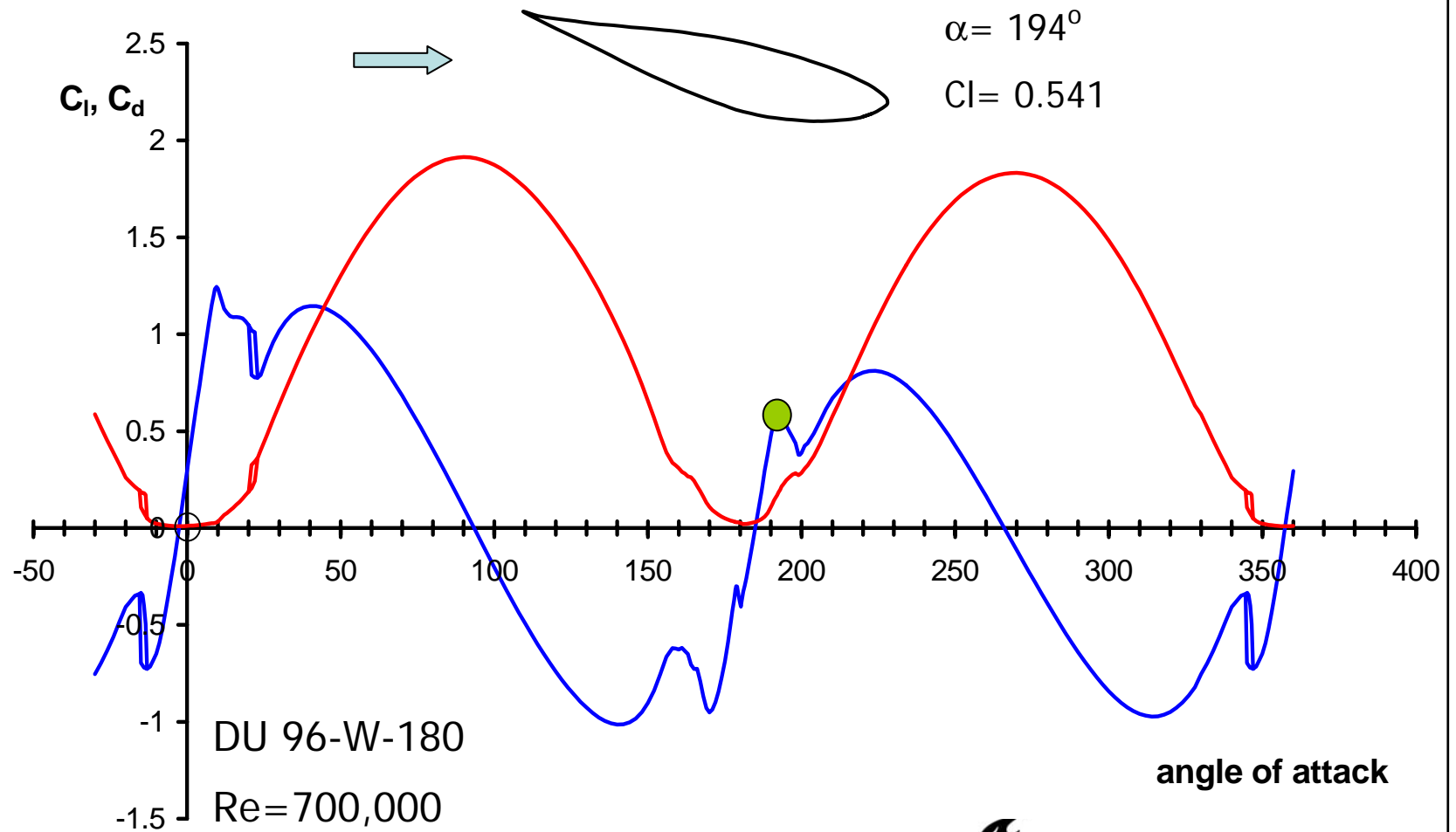
Airfoil testing (360 degrees)



Airfoil testing (360 degrees)



Airfoil testing (360 degrees)

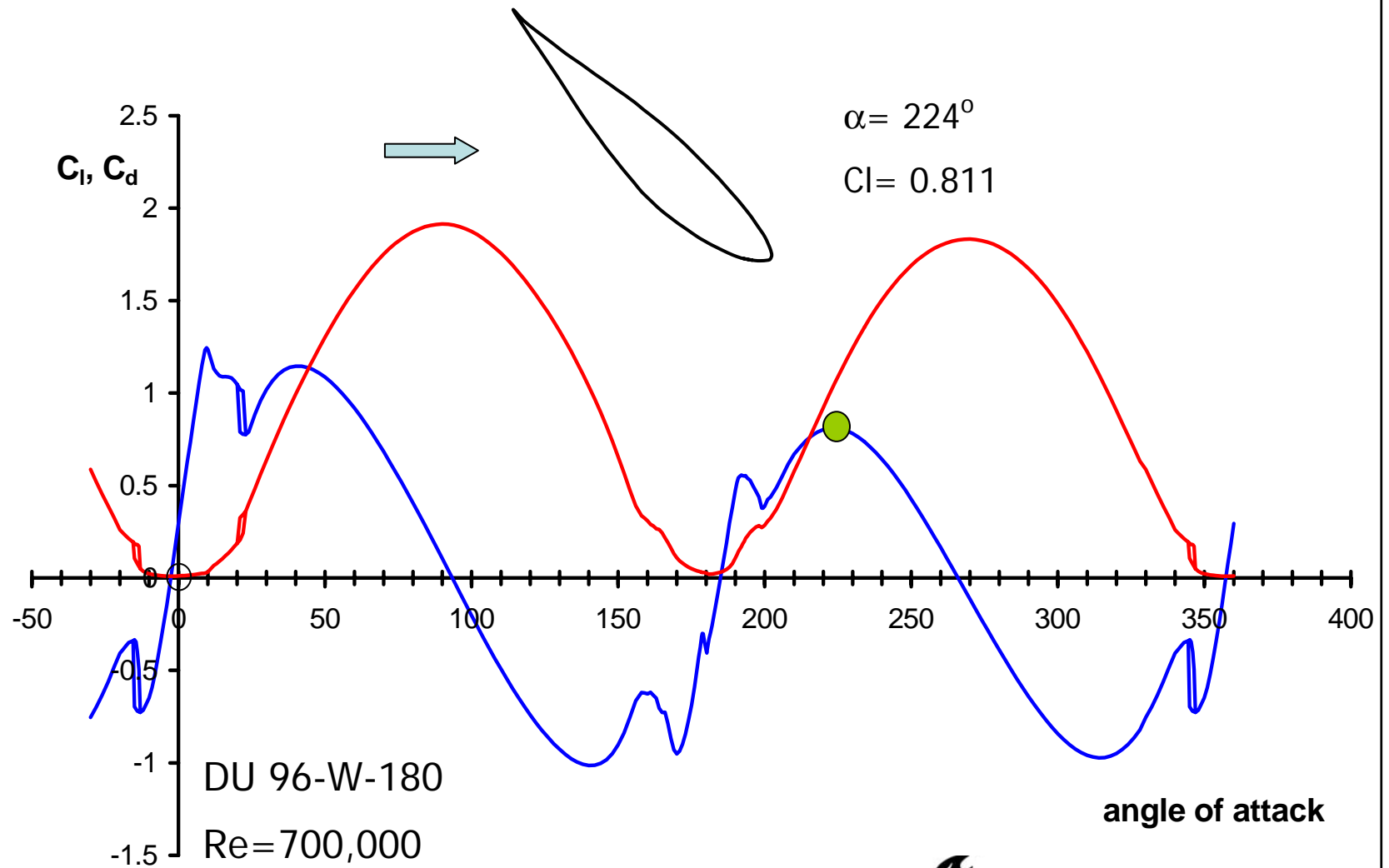


DU 96-W-180

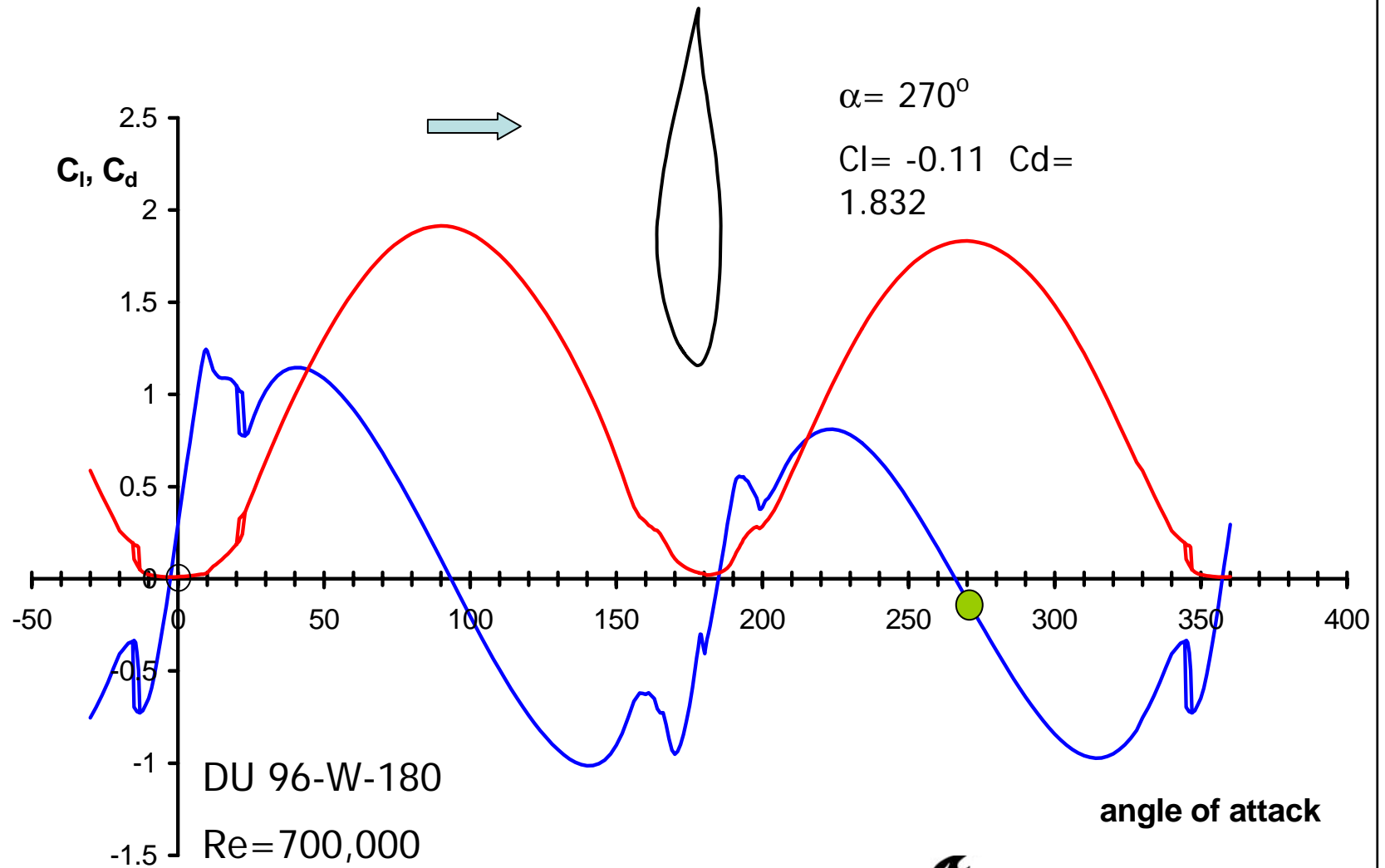
Re=700,000

angle of attack

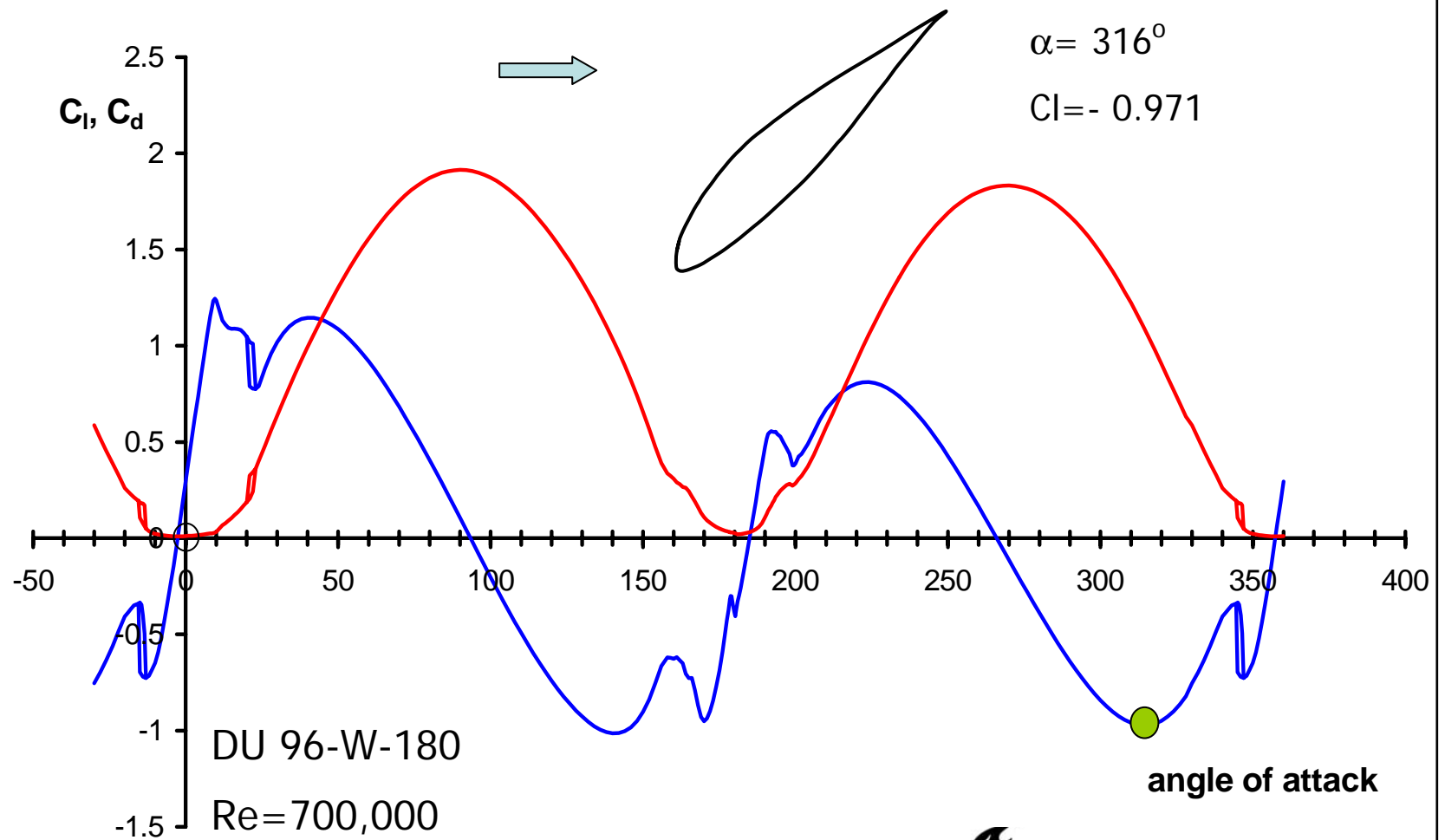
Airfoil testing (360 degrees)



Airfoil testing (360 degrees)



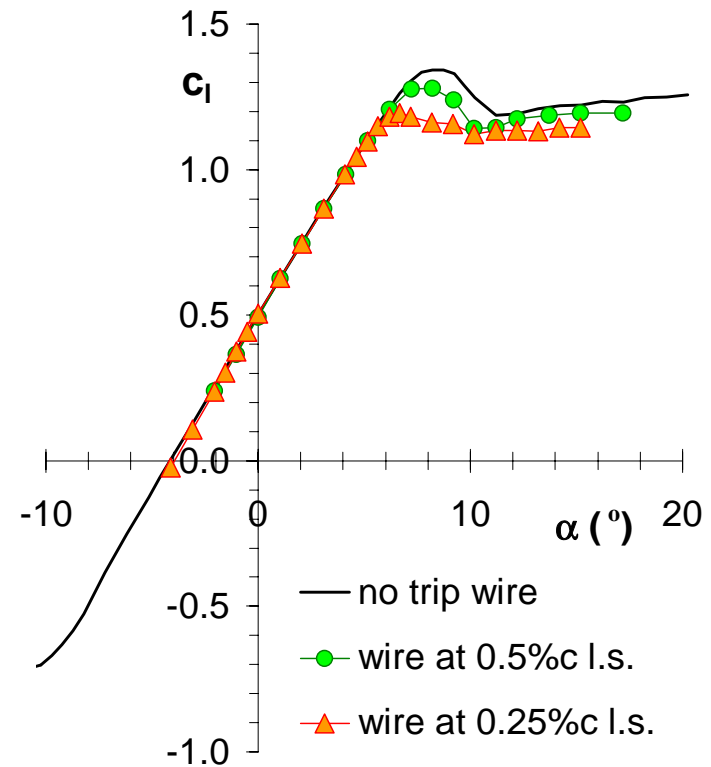
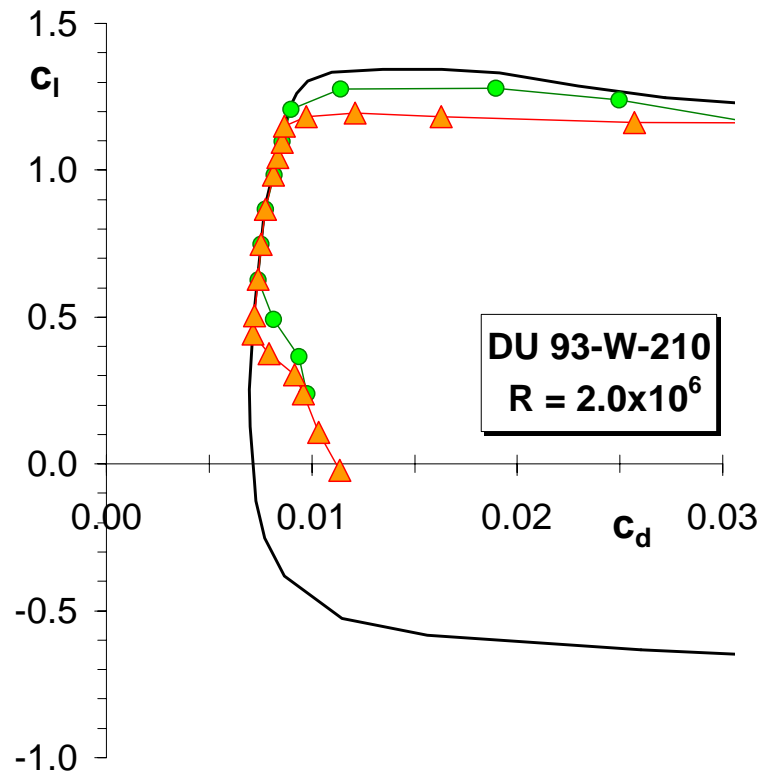
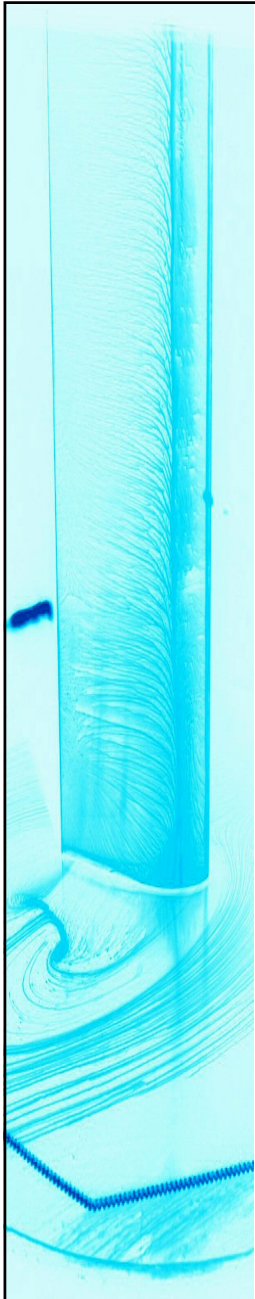
Airfoil testing (360 degrees)



Airfoil testing

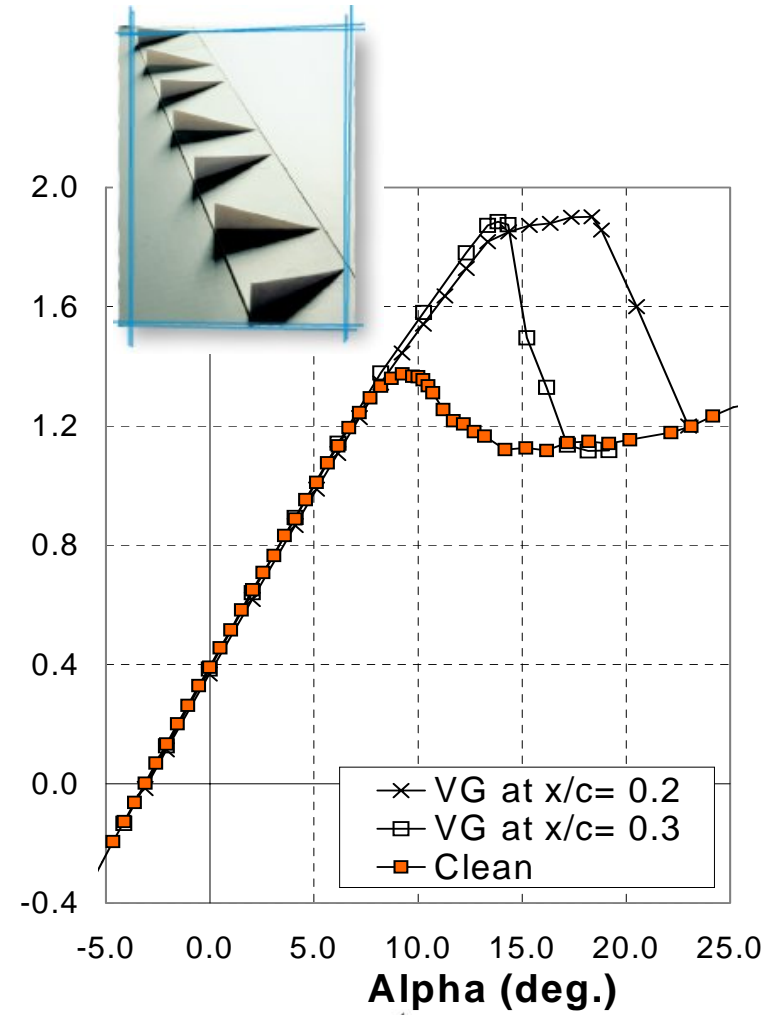
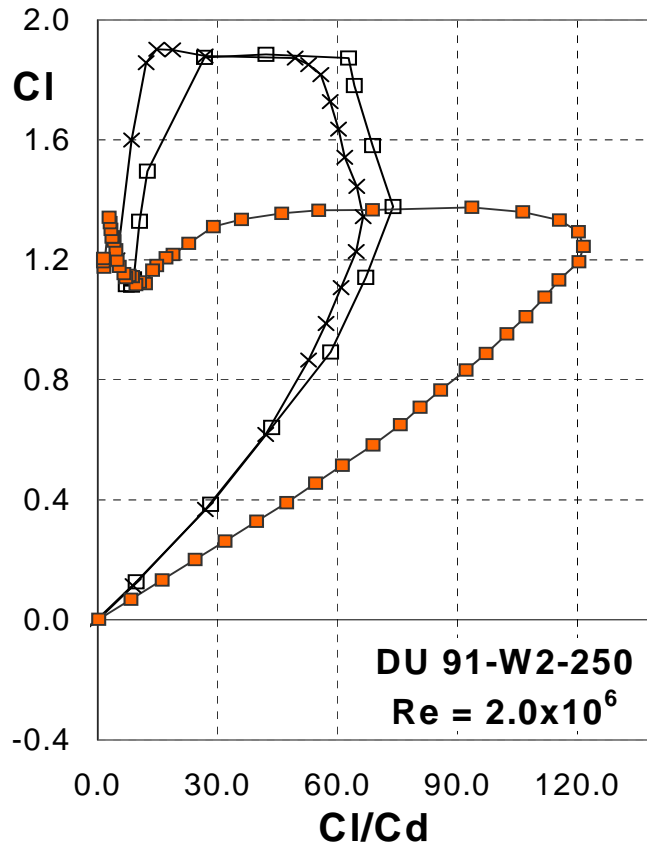
(aerodynamic devices)

- Stall strips



Airfoil testing (aerodynamic devices)

- Vortex generators



Effect on wind turbine performance

(2d stationary performance)

Calculated optimal element performance at mid-span for TSR= 7.5

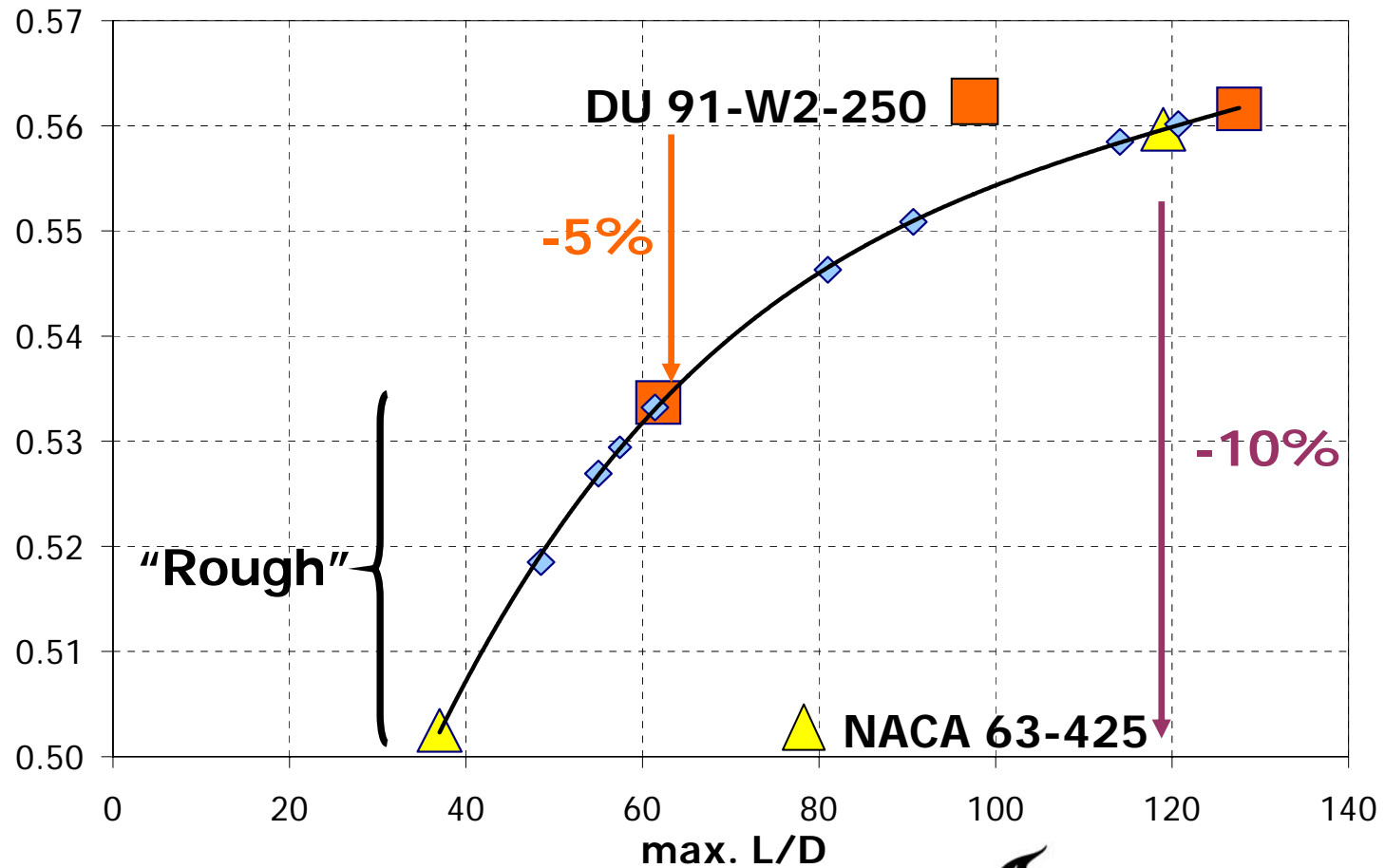
<u>Clean</u>	c/R	L/D-max	"Static load" Cl_max*c	Cp_elem	Loading	Cp
AH 93-W-257	0.106	122	0.149	.56	4%	-0.06%
DU 91-W2-250	0.105	125	0.143	.561	0%	0%
NACA 63-425	0.119	119	0.152	.56	6%	-0.24%
<u>ZZ-tape 5% u.s.</u>						
DU 91-W2-250	0.135	60	.155	.532	8%	-5.1%
NACA 63-425	0.212	39	.212	.503	48%	-10.2%

* "Static load" reference based on 1 year gust for fixed pitch blades

Effect on wind turbine performance (2d stationary performance)

Local Aero Cp

25% thick airfoil class (mid-span for TSR= 7.5)



Overview of HAWT airfoils

General aviation airfoils

- NACA 63-4xx and NACA 63-6xx series
- NACA 64-4xx

Dedicated airfoils

- S8xx series (NREL, USA)
- FFA W-xxx (FOI, Sweden)
- Risø-A1-xxx (also B, P-series, Risø, Denmark)
- DU xx-W-xxx (Delft, Netherlands)

Overview of HAWT airfoils

- Overview of DU-airfoils and users



DU 95-W-180

DU 91-W2-250

DU 96-W-180

DU 97-W-300

DU 00-W-212

DU 00-W-350

DU 93-W-210

GE-Wind, REpower, Dewind, Suzlon, Gamesa, LM Glasfiber, NOI Rotortechnik, Fuhrlander, Pfeiderer, EUROS, NEG Micon, Umoe blades, Ecotecnia

