

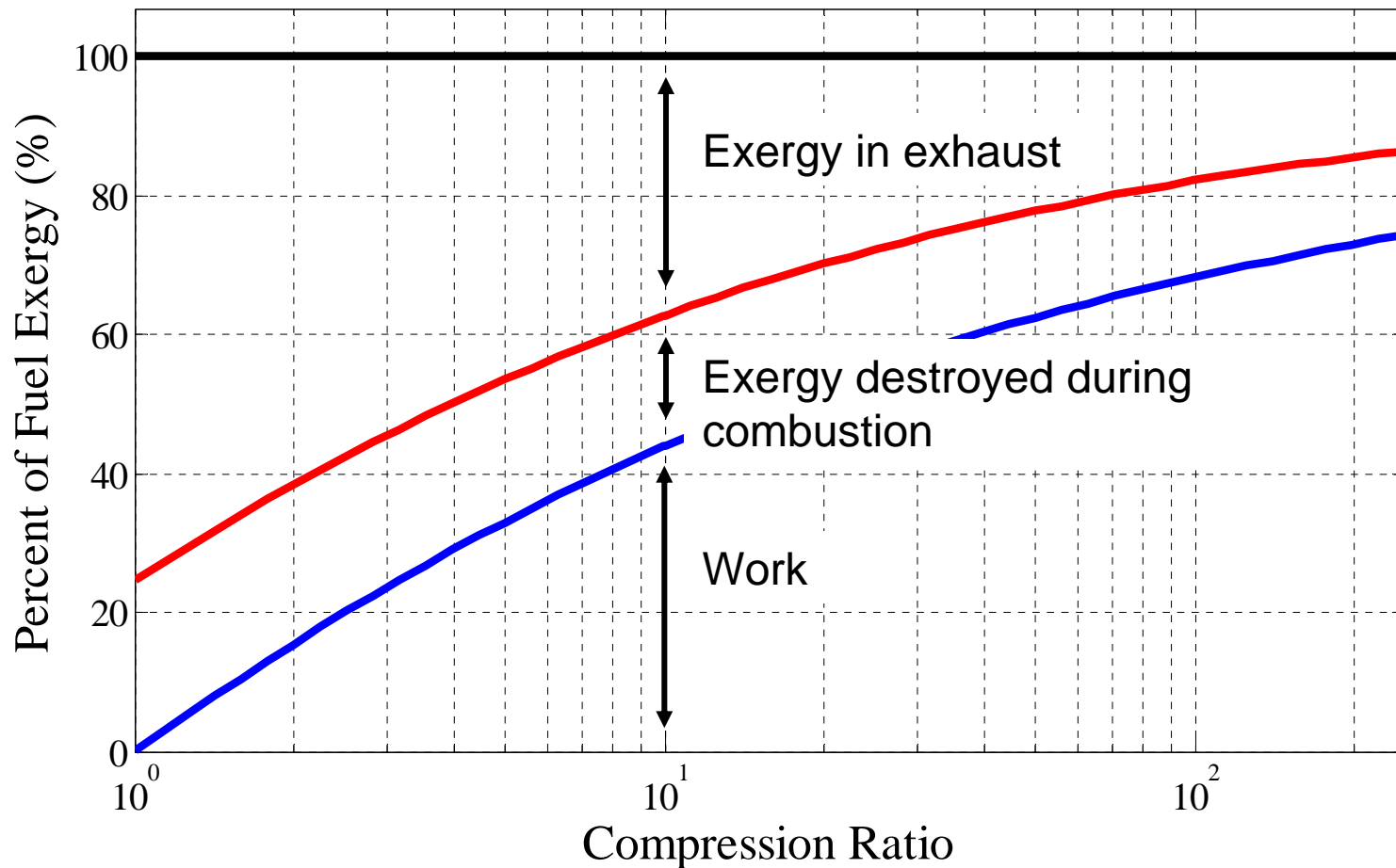
The background features a large, semi-transparent watermark of the Stanford University seal. The seal is circular and contains a tree in the center, surrounded by the text 'STANFORD UNIVERSITY' and 'DIE ZEIT DER FREIHEIT'. The year '1891' is visible at the bottom of the seal.

# Development of Low-Exergy-Loss, High-Efficiency Chemical Engines

Chris Edwards,  
Shannon Miller, Matthew Svrcek,  
Sanakaran Ramakrishnan, Kwee Yan Teh,  
Olivier Lacroix, Joe Wilson

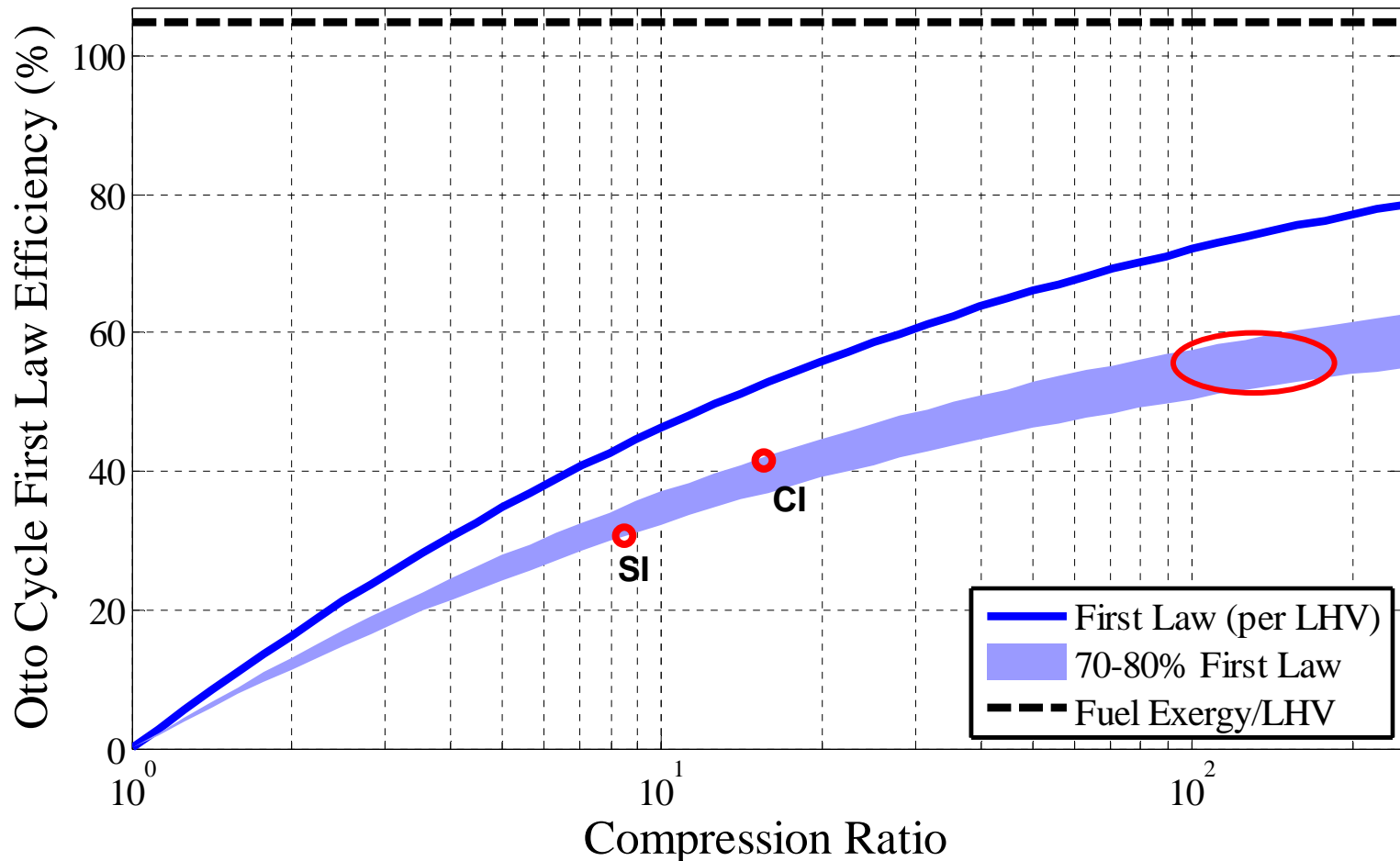
Department of Mechanical Engineering  
Stanford University

# Fuel Exergy Distribution for Reactive Otto Cycle



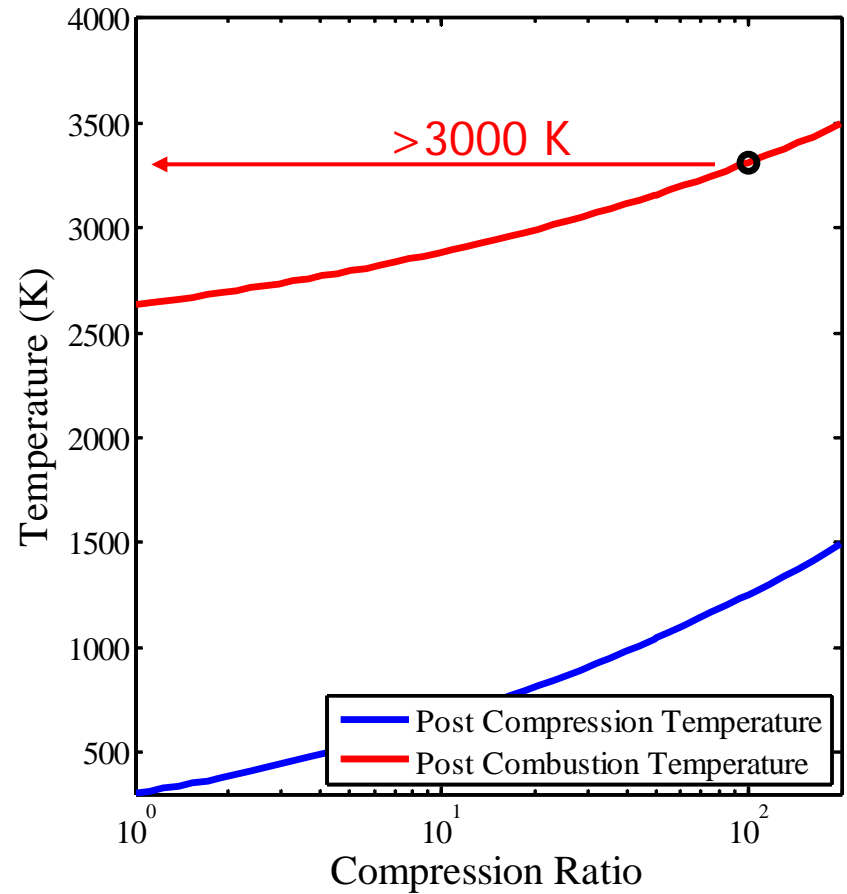
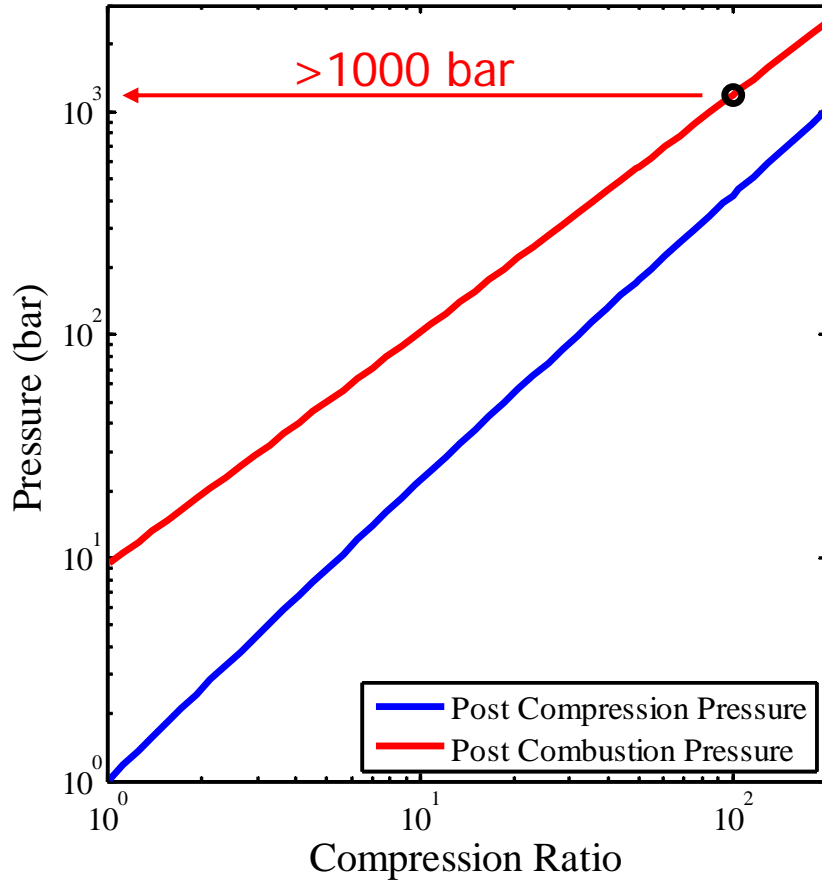
Stoichiometric propane/air modeled as ideal gases.

# What fraction of theoretical efficiency might be achieved?



Stoichiometric propane/air modeled as ideal gases.

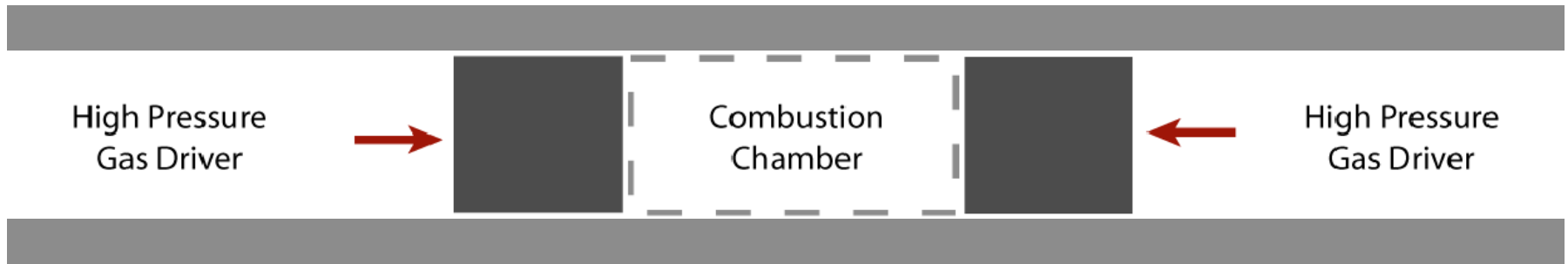
# Extreme Compression Conditions



Stoichiometric propane/air mixture modeled as ideal gases.

Adiabatic, isentropic compression followed by adiabatic constant volume combustion

# Achieving Extreme Compression

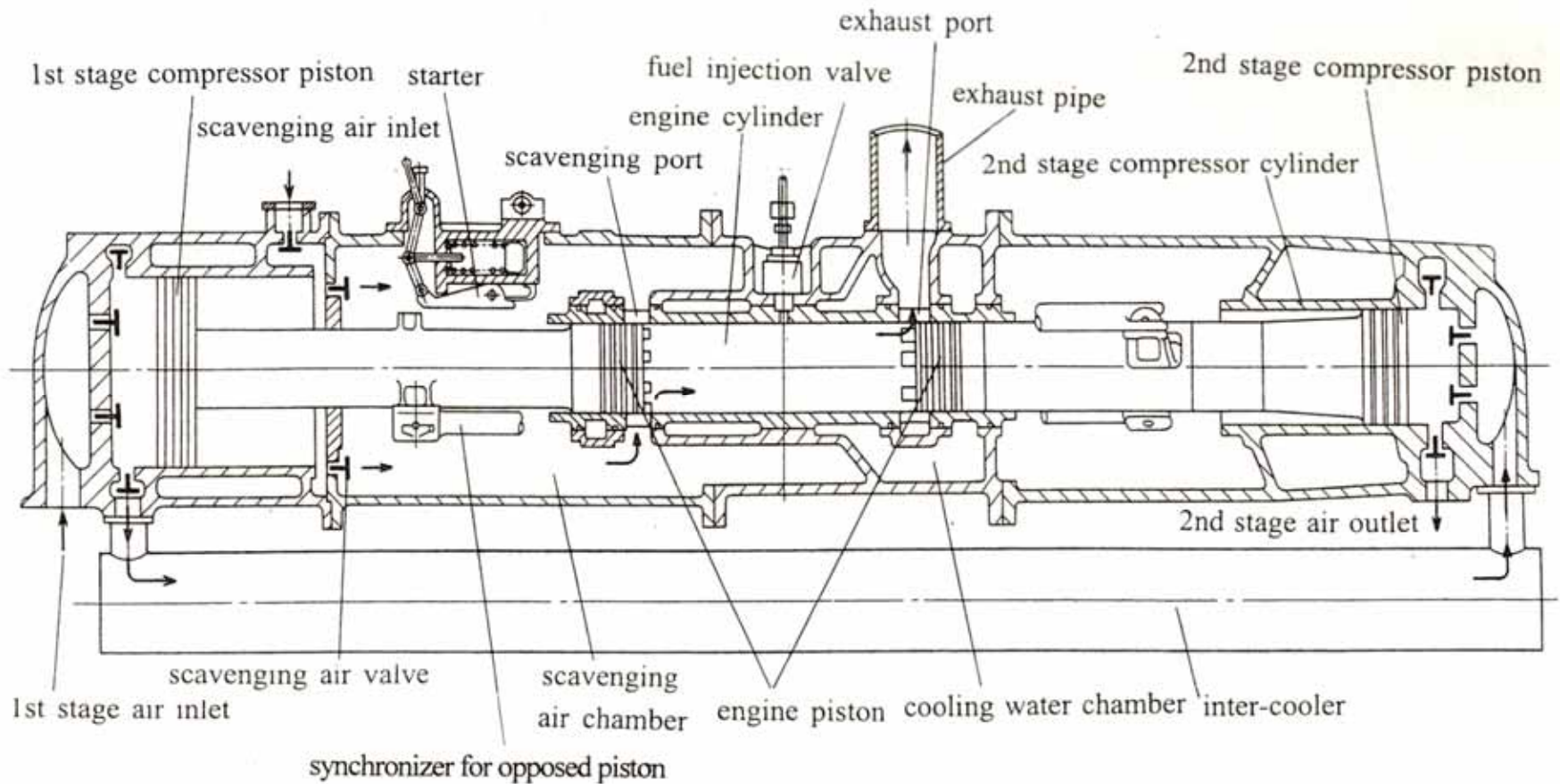


## Basic Requirements:

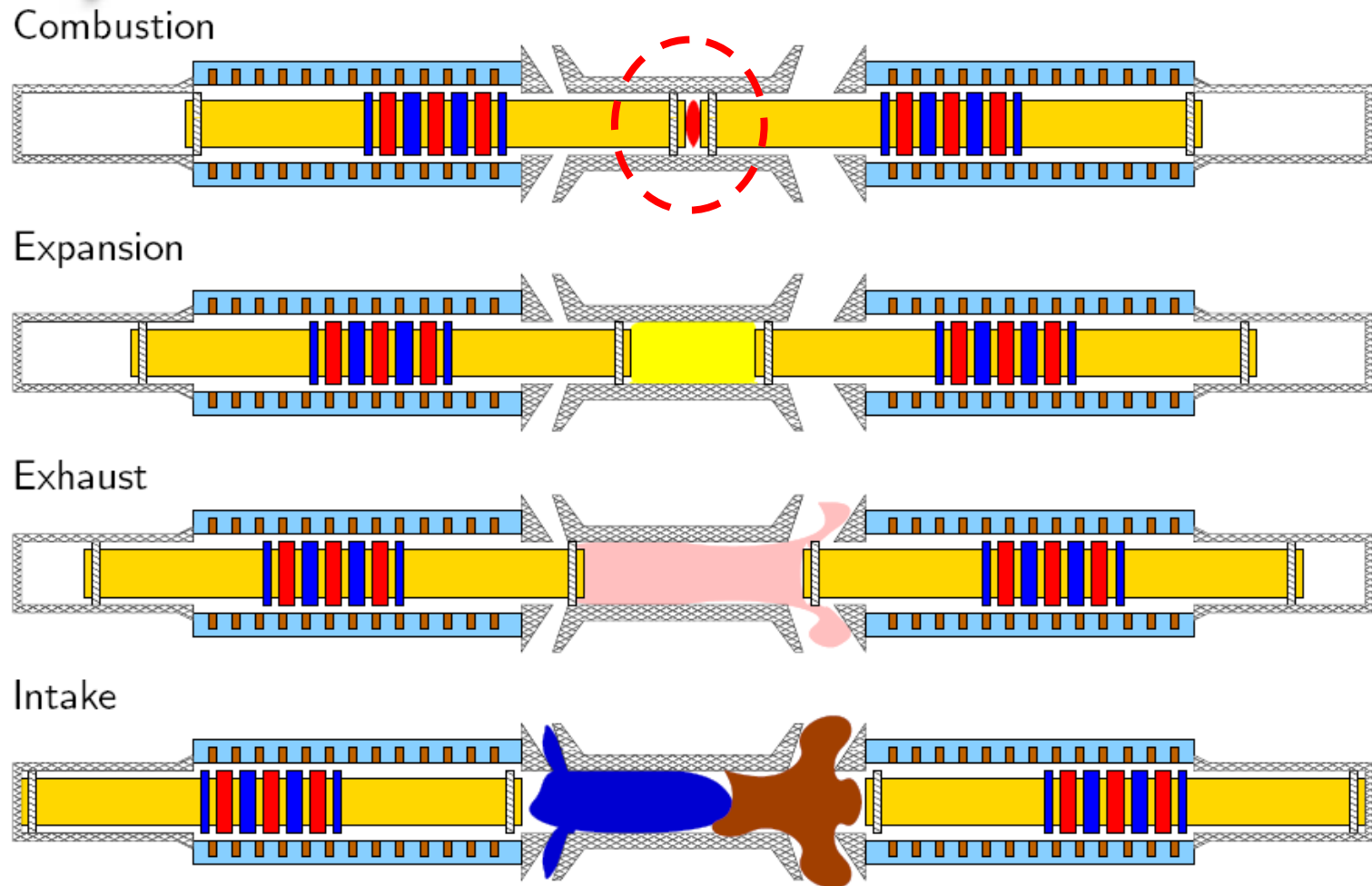
- High compression ratio (100:1 and higher)
- Approach for reducing heat transfer
  - Large TDC volume to reduce heat transfer
  - High piston speeds to reduce time available for heat transfer
- Approach for high forces
  - Free pistons
  - Multiple pistons

**Design apparatus for testing the Critical Questions**

# Junkers Free-Piston Compressor



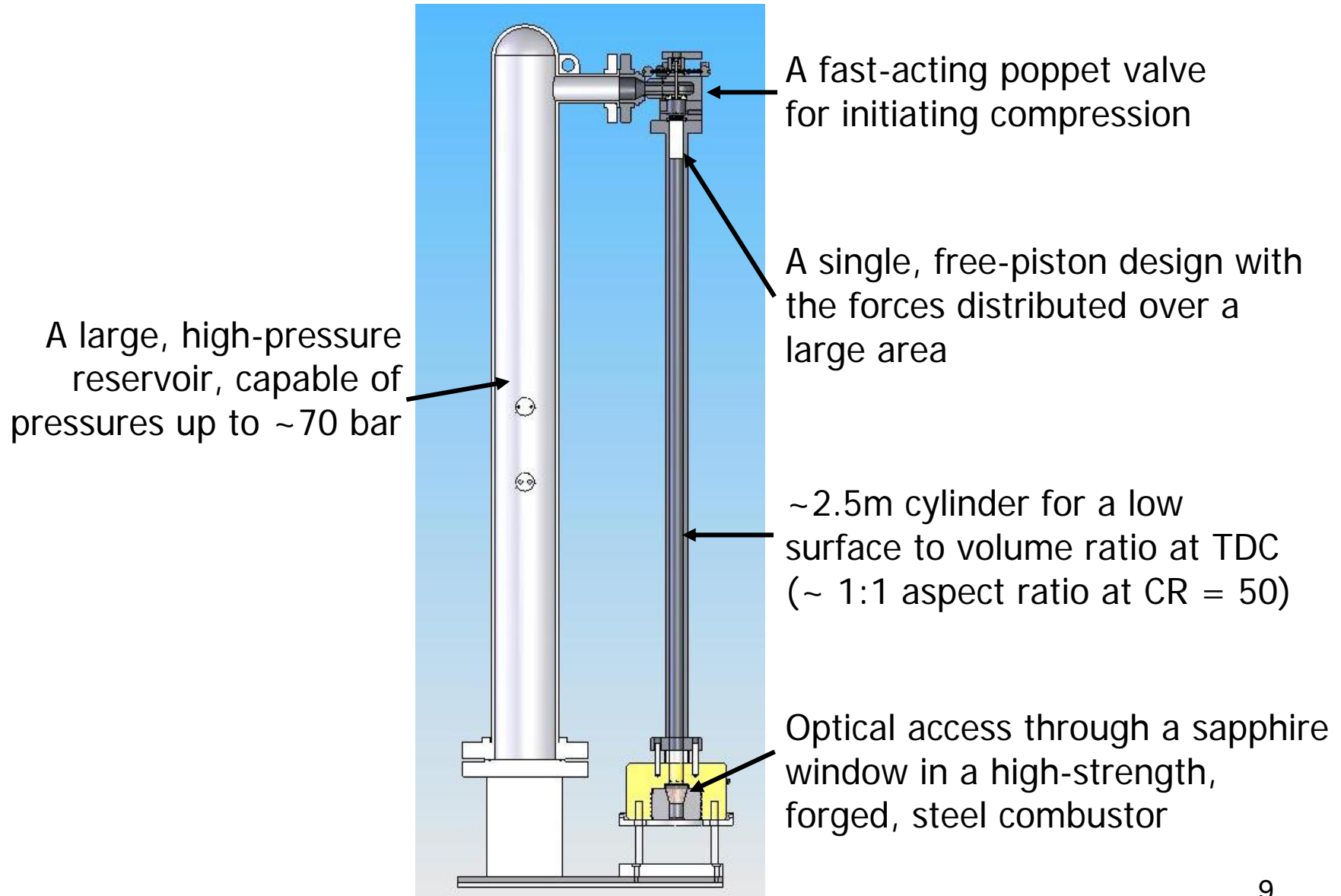
# Van Blarigan/Aichlmayr Linear Alternator Concept



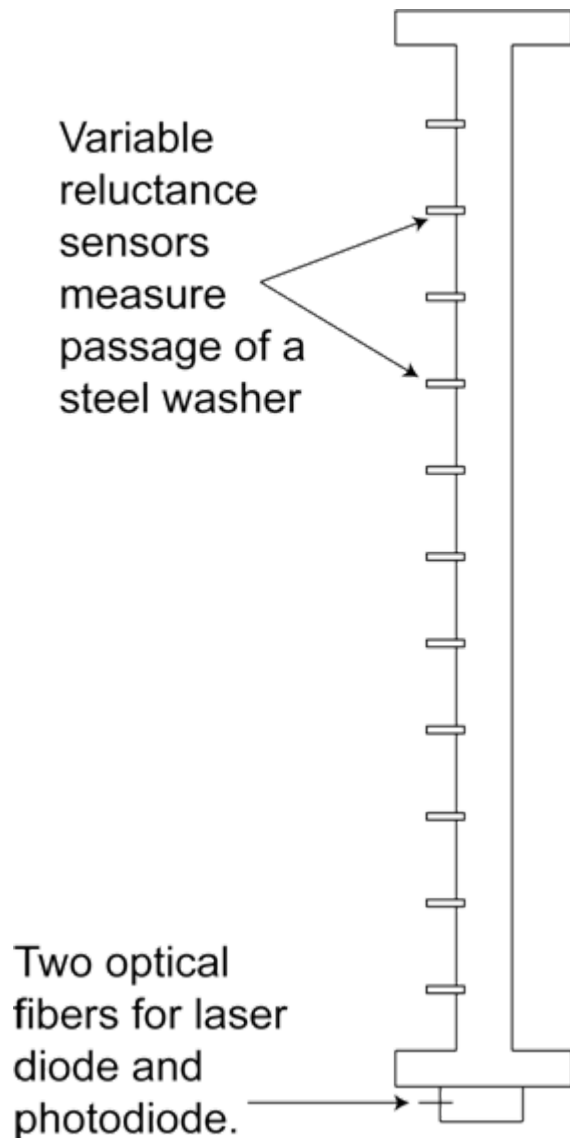
# Critical Questions

- 1) Can an engine handle the high temperatures and pressures required for extreme-state combustion?
  - 1) Material stress
  - 2) Wall temperatures and heat transfer
  - 3) NO<sub>x</sub>
- 2) Can the rings withstand the pressure and piston speed?
  - 1) Seal survivability
  - 2) Sealing ability: Can we achieve <1% blowby?
- 3) Can we implement and control combustion?
  - 1) Ignition phasing
  - 2) Combustion control
  - 3) Rate and efficiency

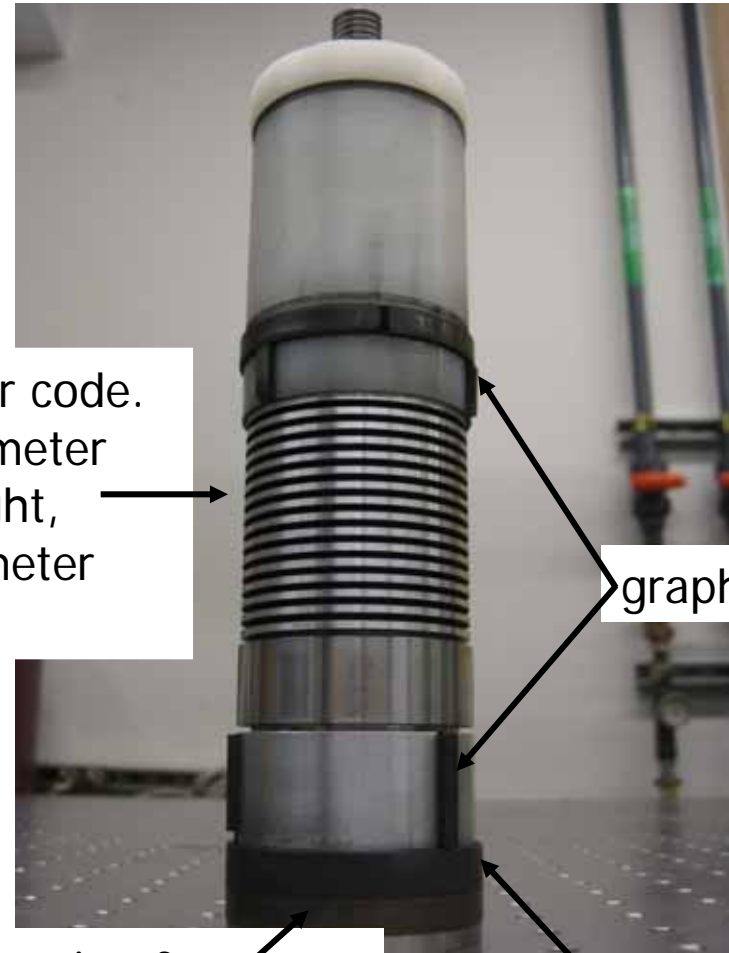
# Experimental Design



# Piston Position Sensing



## Current Piston Design



Optical bar code.  
Outer diameter reflects light,  
inner diameter does not.

graphite skins

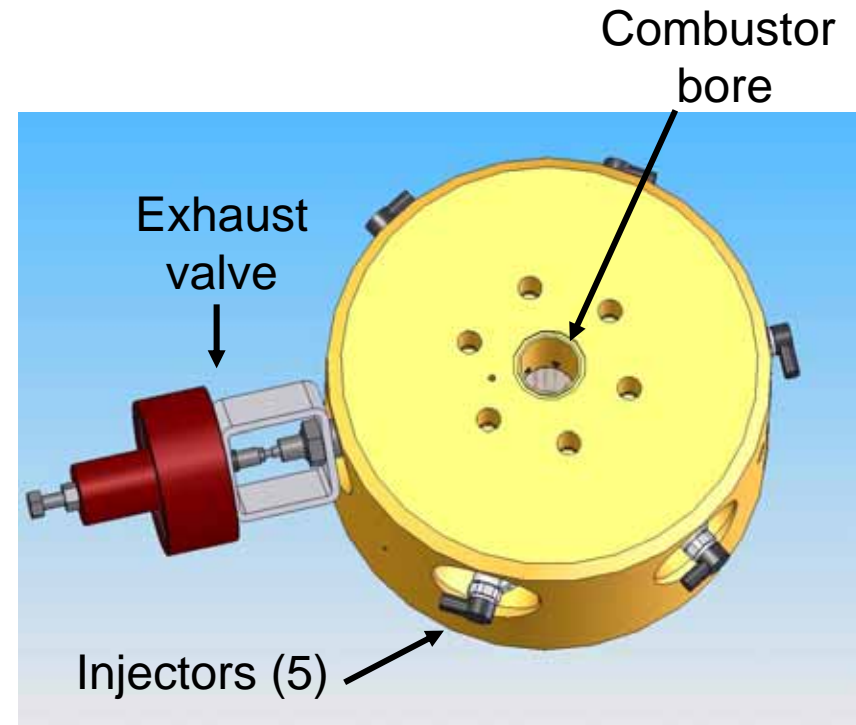
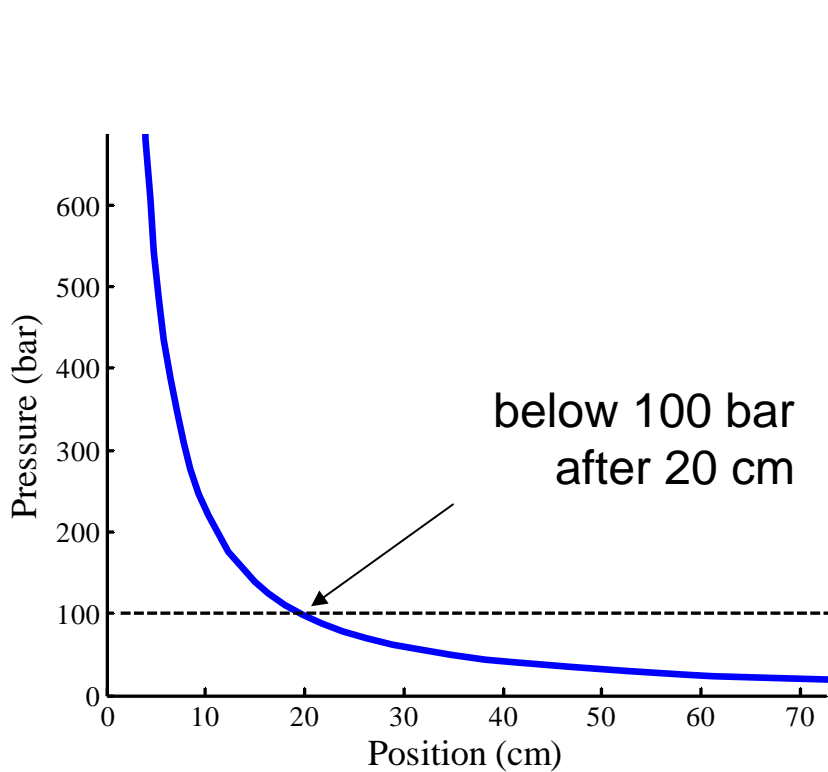
copper ring for  
thermally protecting  
the rings

graphite ring

# Combustor Design

Material stress strategy: Use pressure profile to our advantage

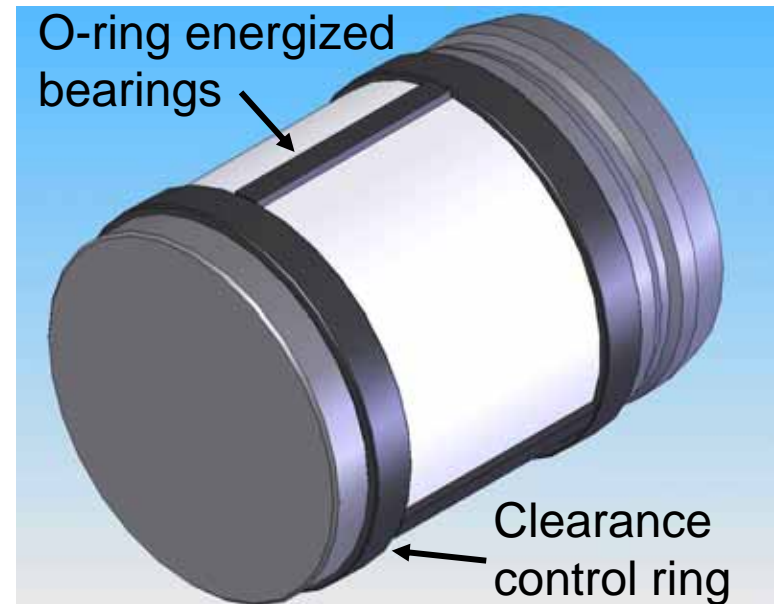
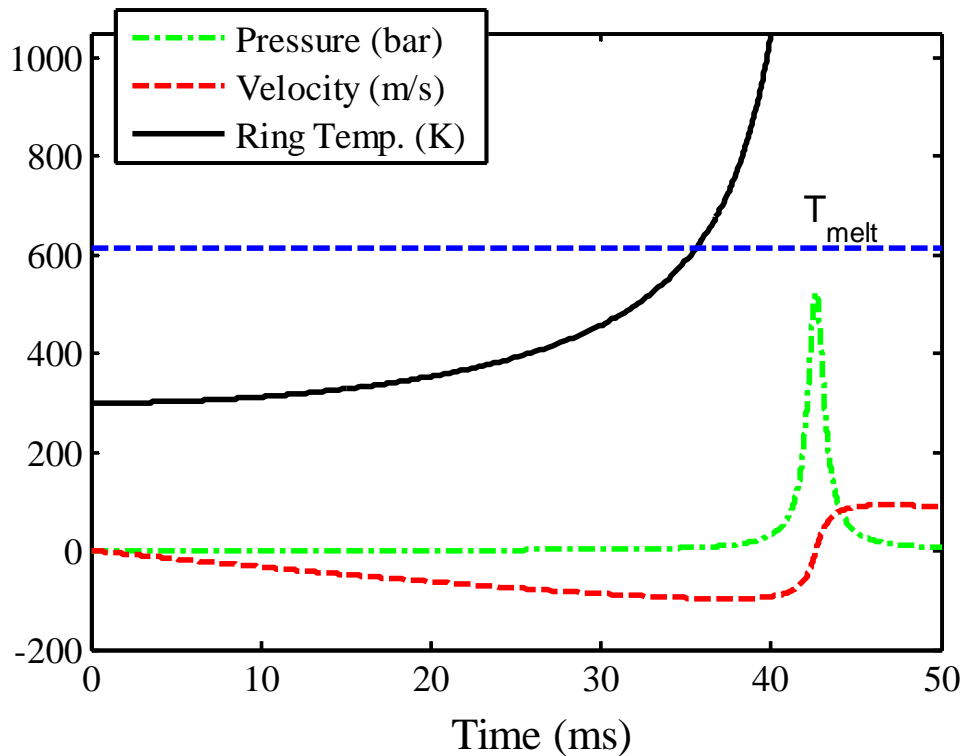
Injection strategy: Five Bosch high-pressure diesel injectors



Combustor with 2000 bar working pressure

# Piston Design

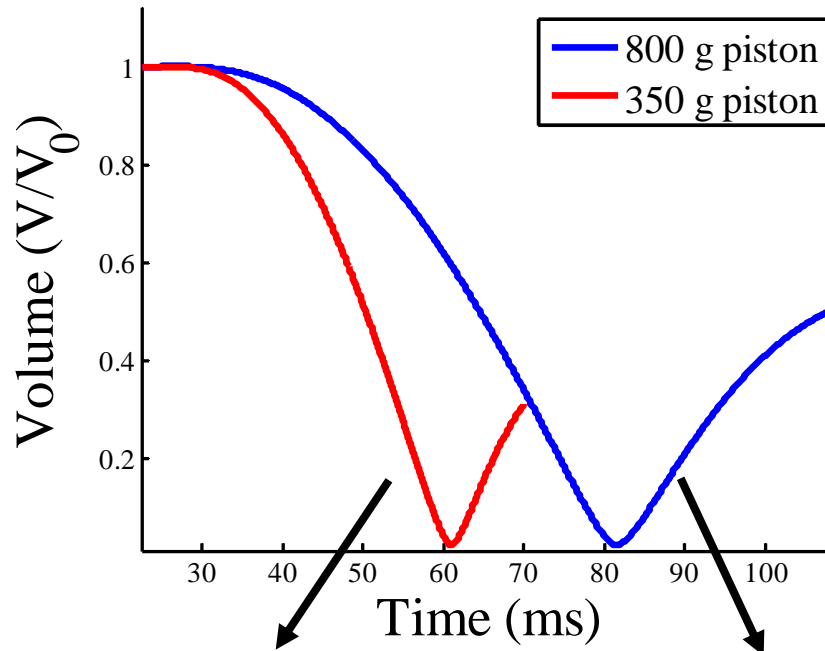
Sealing strategy: Use clearance control, rather than conventionally energized rings to reduce ring temperatures



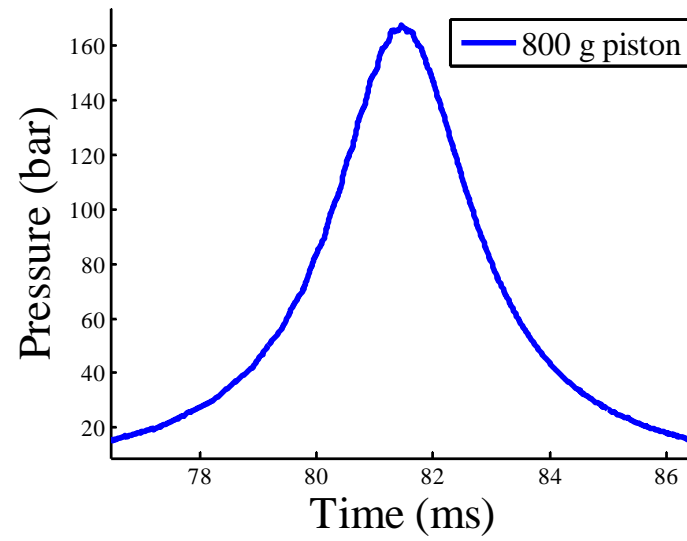
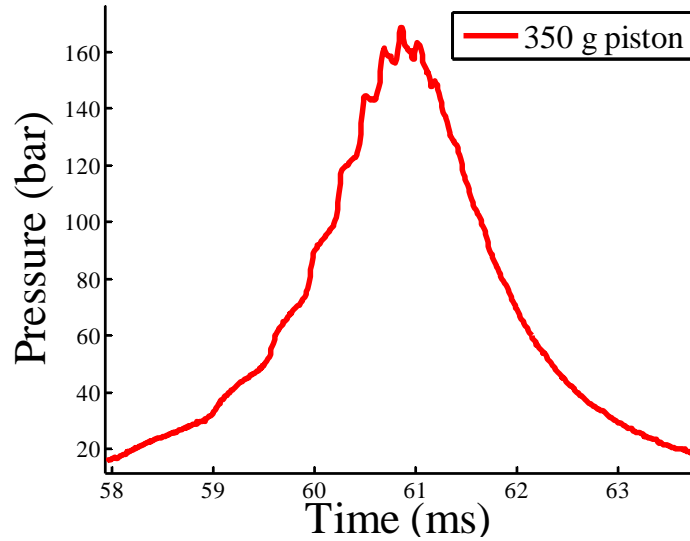
This low friction strategy also gives high experimental repeatability.

# Air Compression: Initial Findings

MPS = 90 m/s  
CR = 45

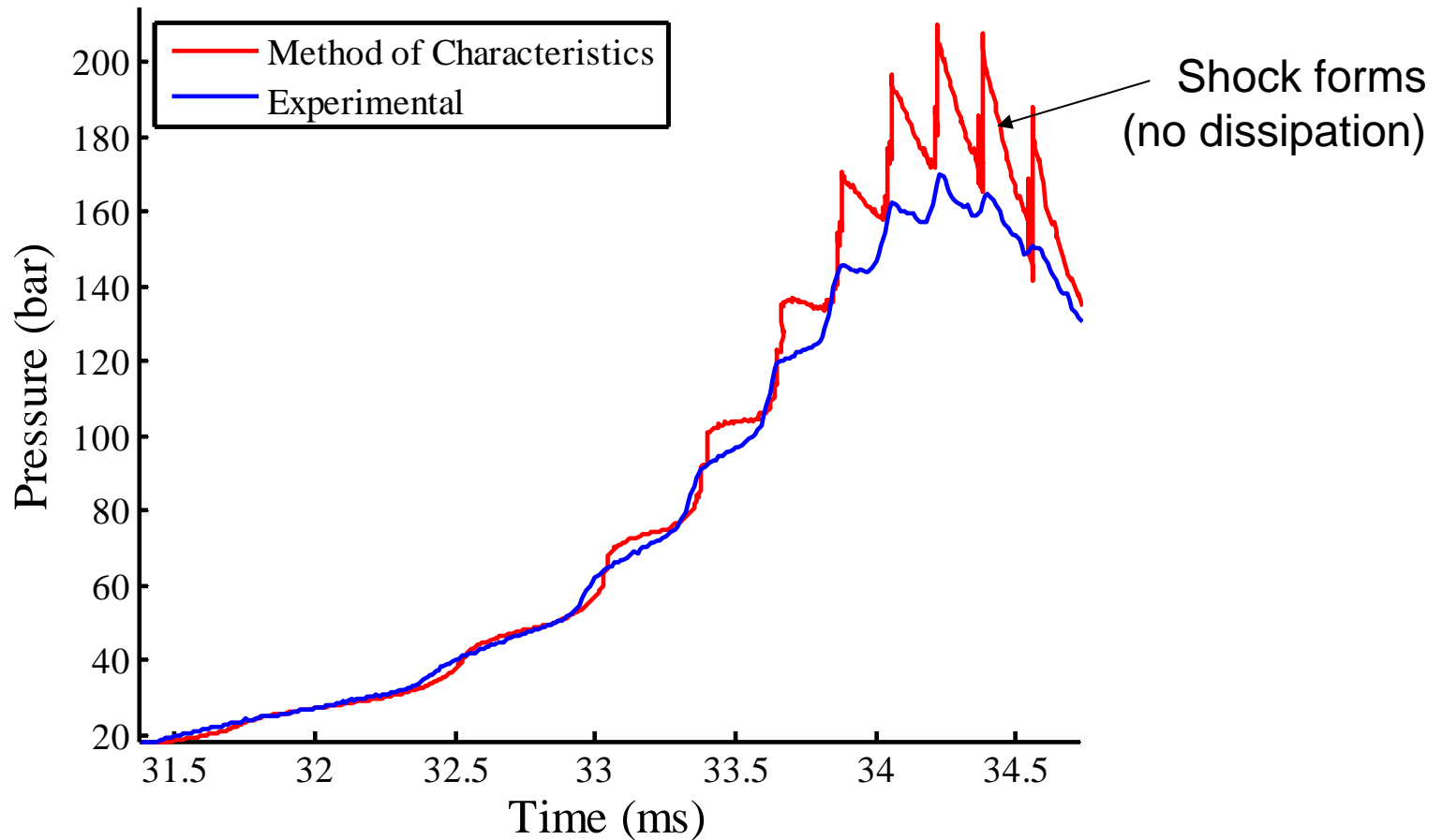


MPS = 60 m/s  
CR = 45

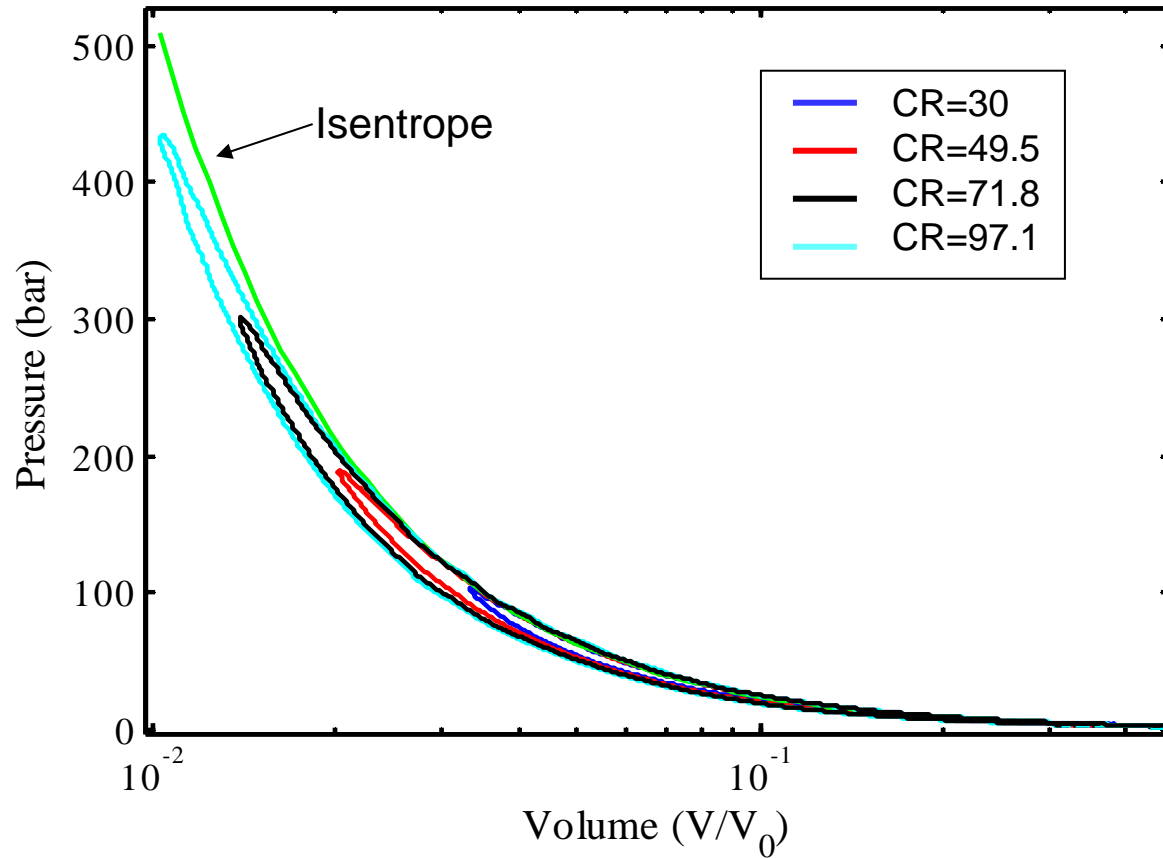


# Acoustic Waves

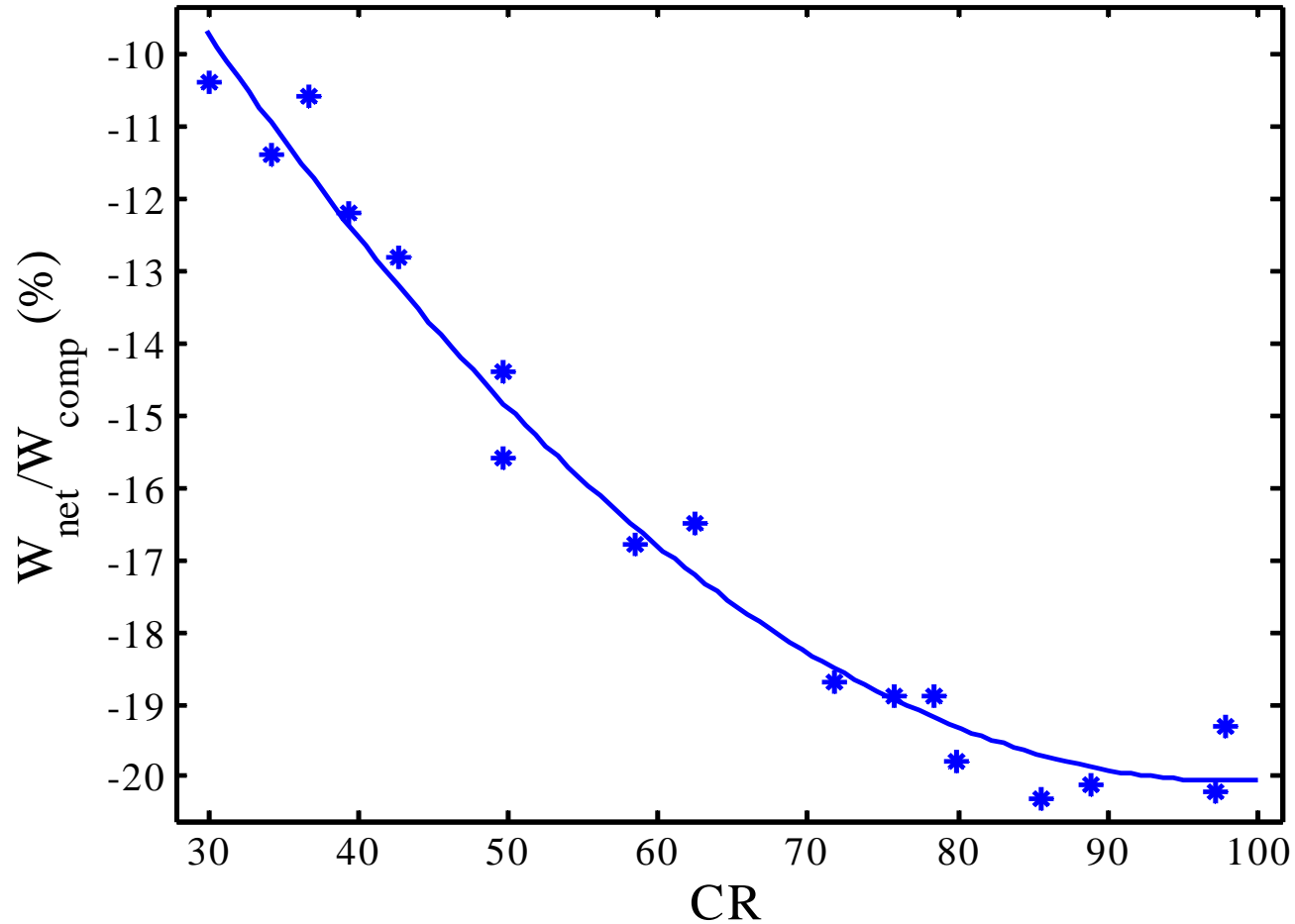
Method of characteristics simulations show that high piston accelerations cause acoustic waves.



# P-V Plots: Air Data

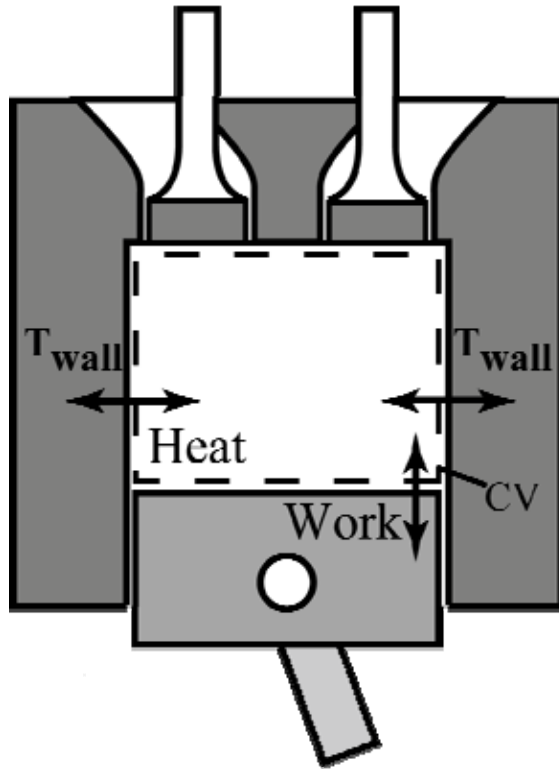


# Total Losses Over an Air Cycle



Losses consist of heat and mass transfer.

# Heat Loss on Compression



## Model Assumptions

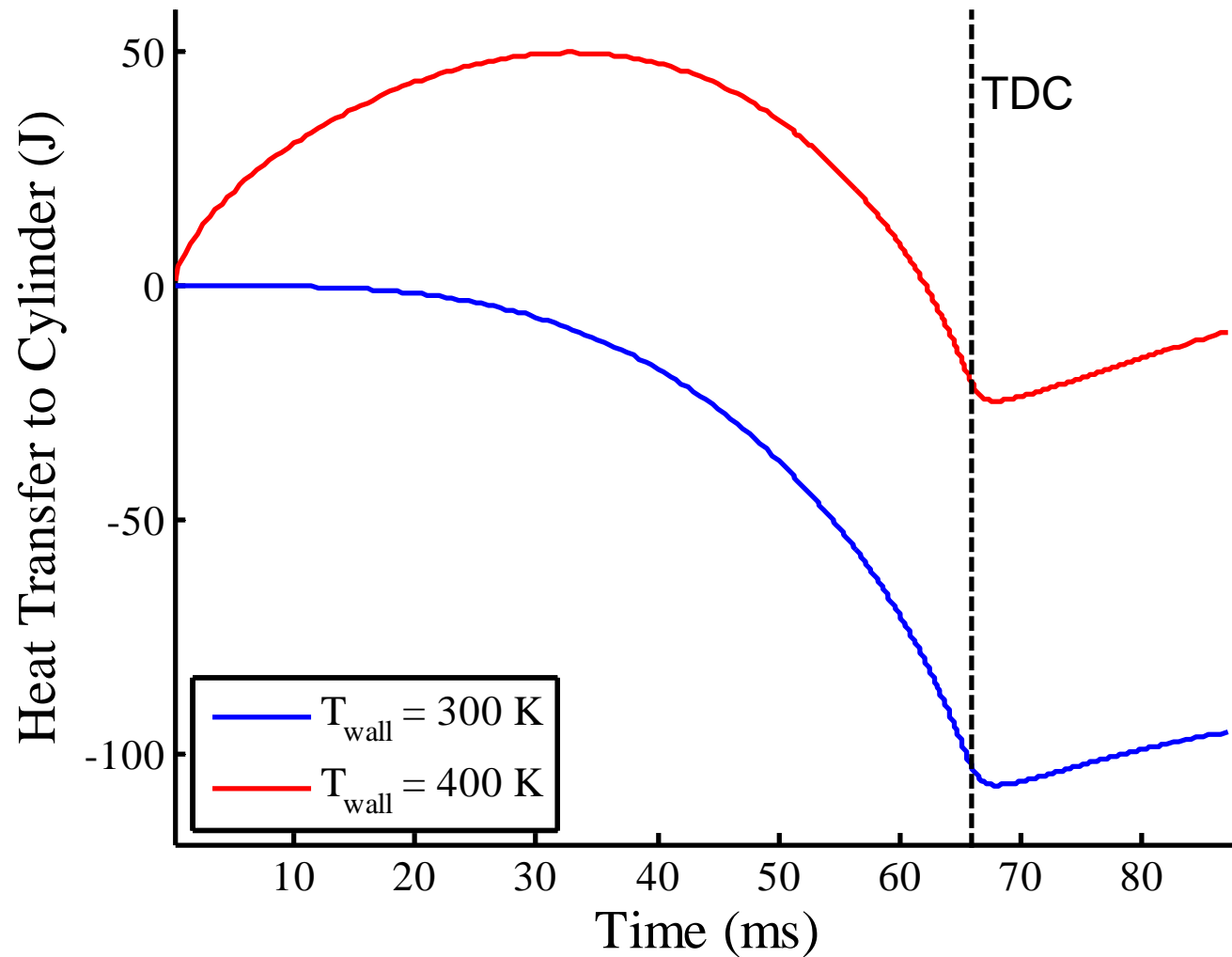
- simulates boundary layer formation assuming conduction as only form of heat transfer
- one dimensional
- constant wall temperature
- uniform pressure

Modeled heat transfer losses for CR = 50 air data:

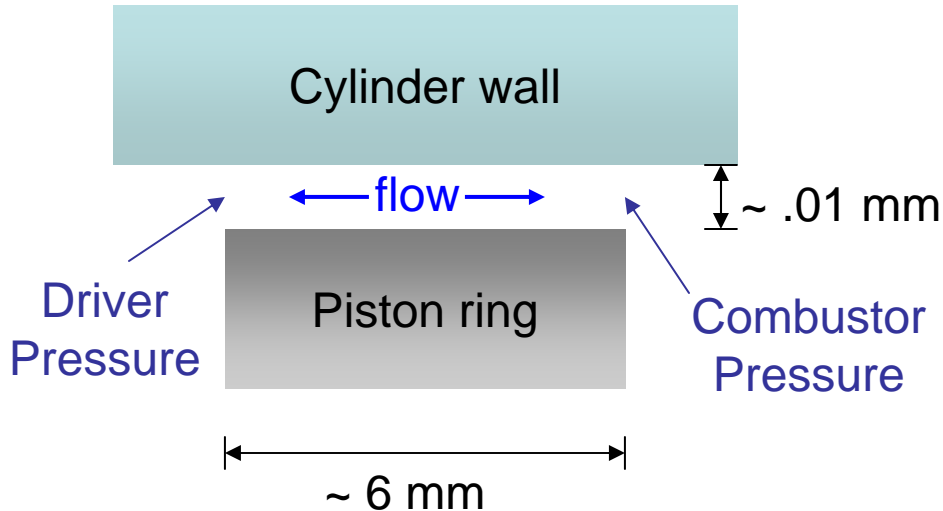
300 K → ~2% compression work lost to HT  
~3% drop in peak pressure from isentropic

400 K → ~0.2% compression work lost to HT  
negligible pressure drop from isentropic

# Heat Transfer to Cylinder



# Mass Loss Past Rings



## Model Assumptions

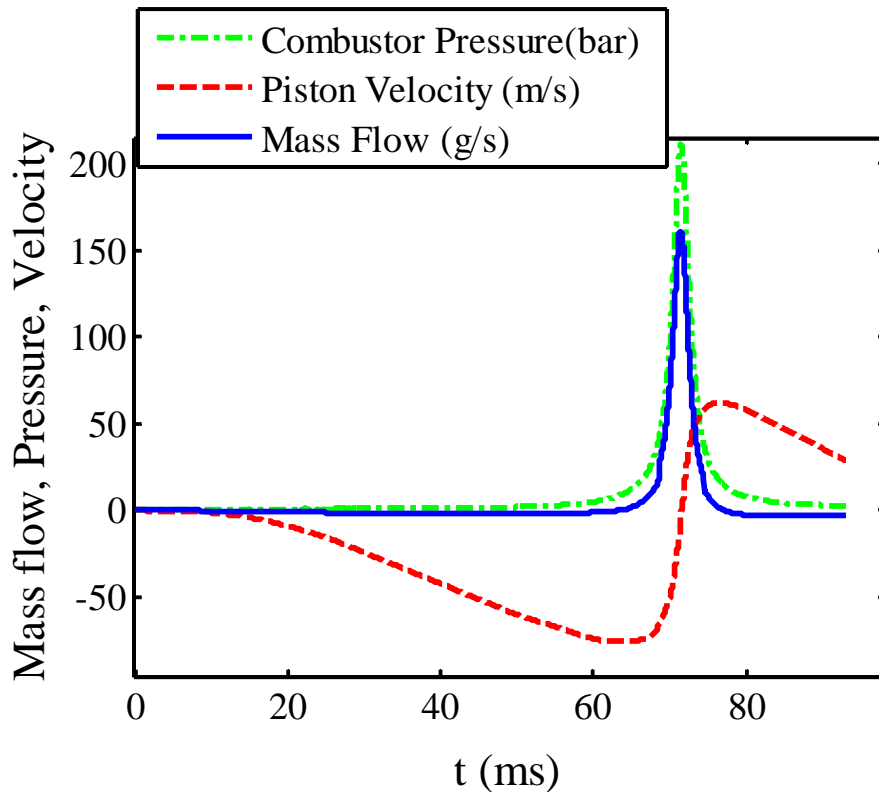
- 1D, isothermal flow through ring-wall gap driven by pressure drop and piston motion
- Viscous effects included via a friction factor
- Pressure and piston motion from experimental runs

Modeled losses for the CR = 50 air data:

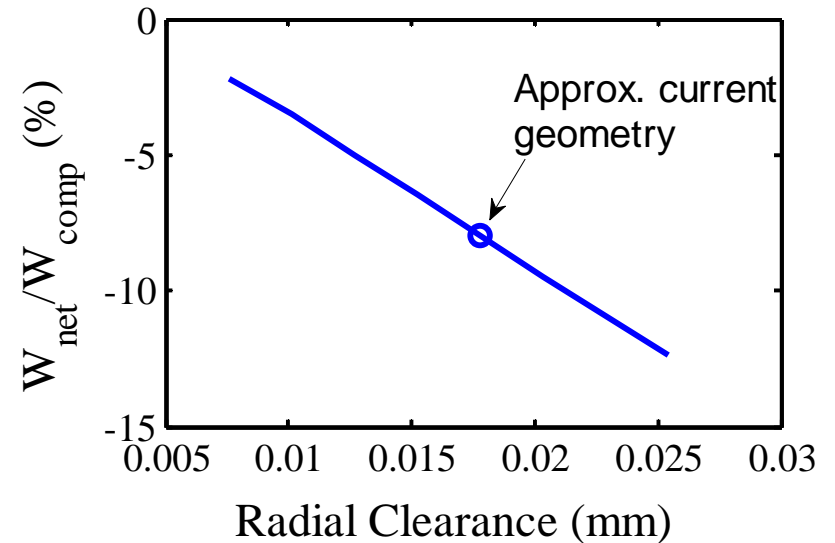
8% of compression work ( $W_{\text{net}} / W_{\text{comp}}$ )  
4% of peak pressure ( $P / P_S$ )

# Mass Transfer to Cylinder

Mass flow profile for CR = 50



Effect of clearance gap on work loss for CR = 50

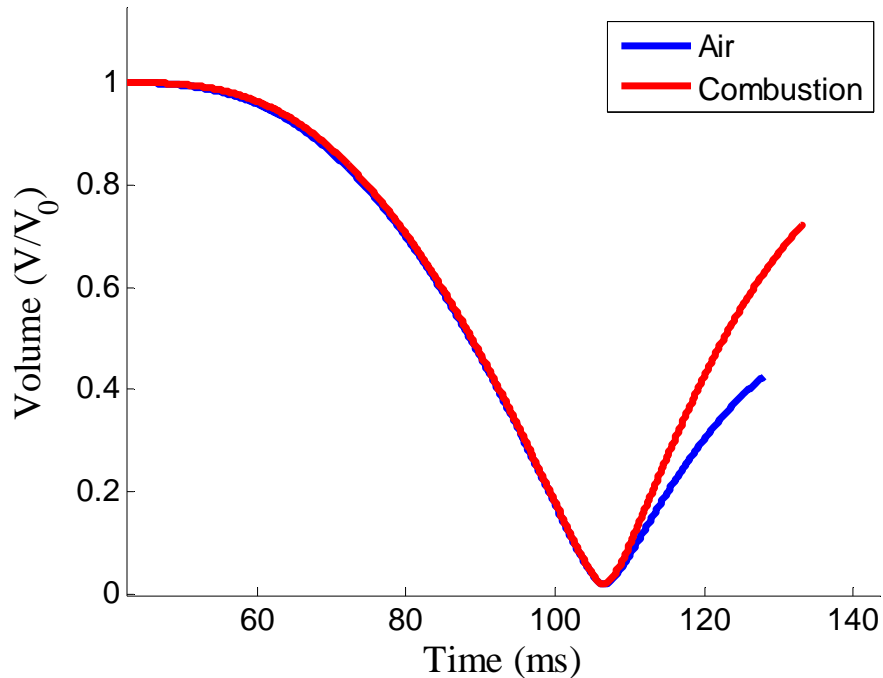


Radial bore tolerances:

combustor  $\pm .0015$  mm

cylinder  $\pm .007$  mm

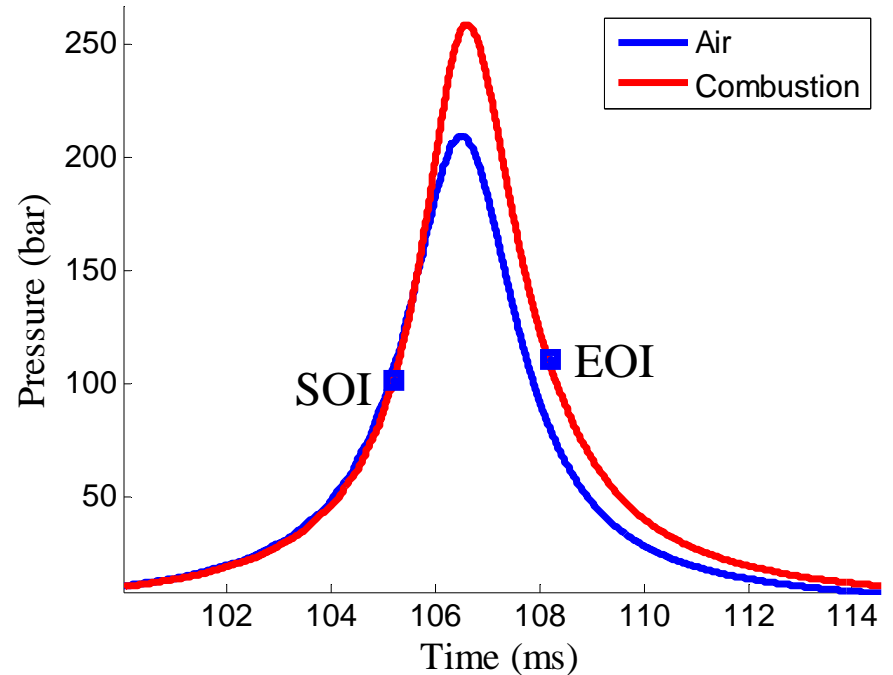
# Combustion Data: CR = 50



MPS ~ 40 m/s

Peak Speed ~ 75 m/s

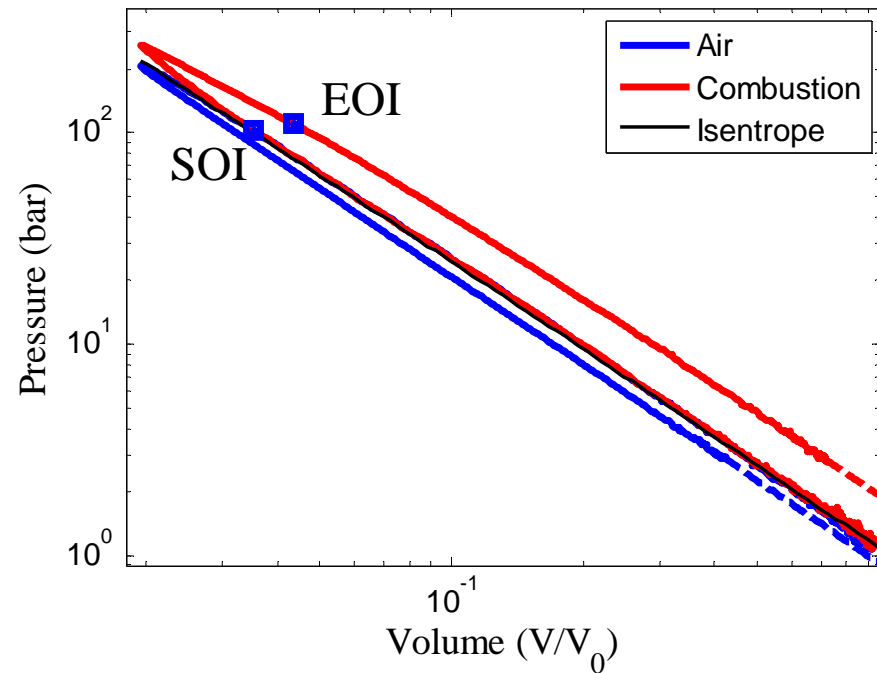
Piston mass = 1.03 kg



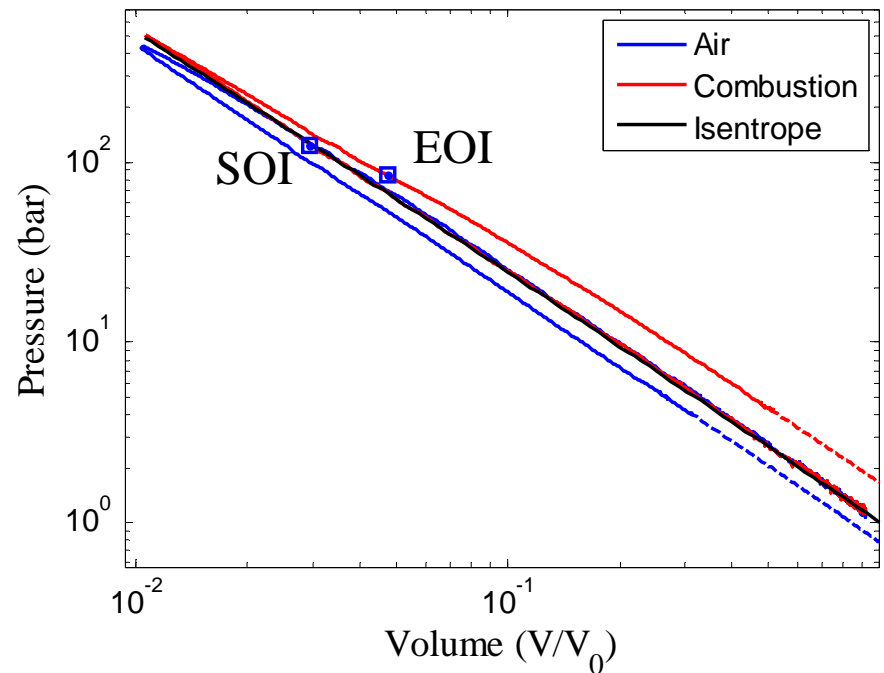
CR = 51.4

Driver Pressure = 10.2 bar

# Comparison to Higher Compression

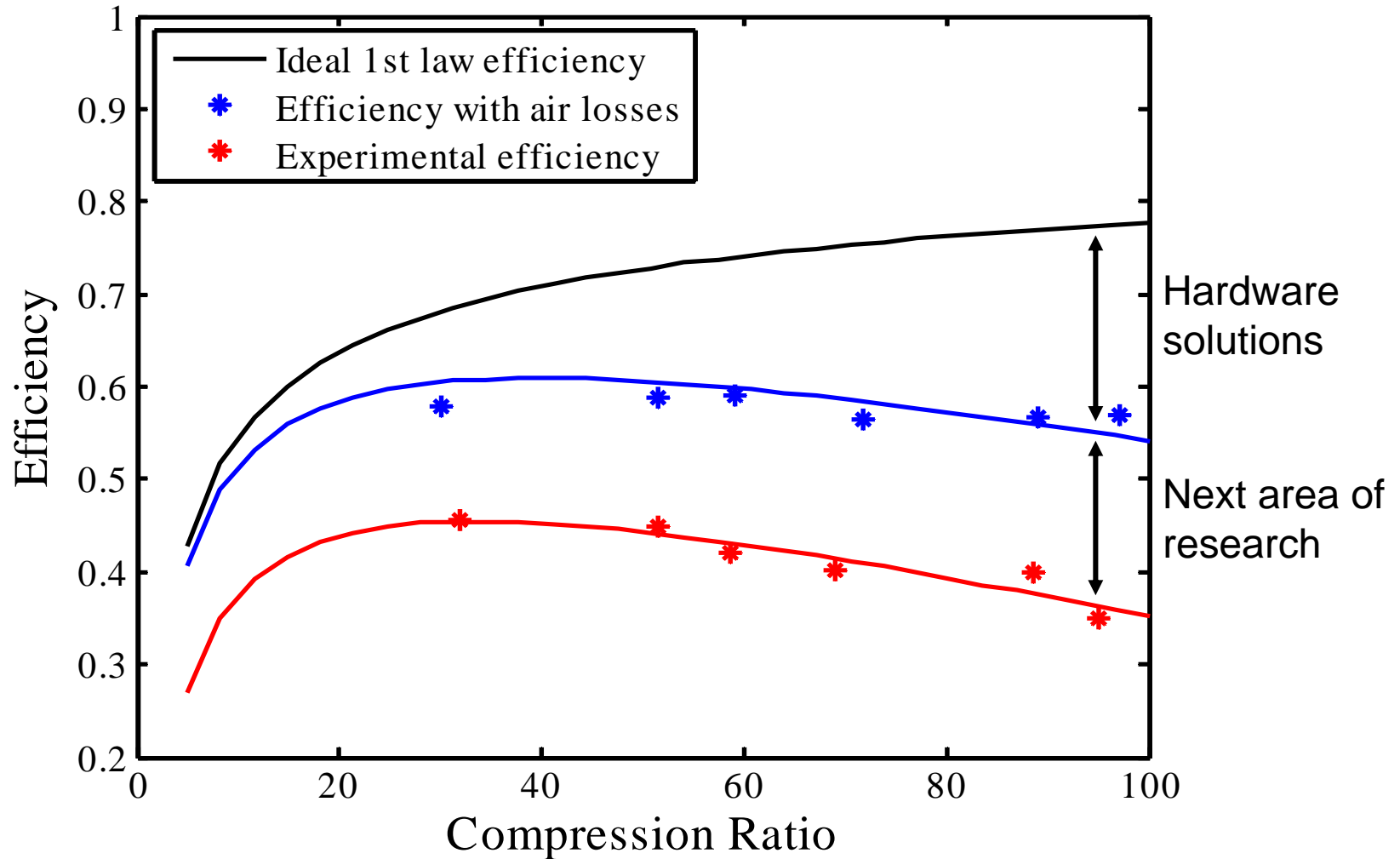


CR = 51.4 (Combustion)



CR = 95 (Combustion)

# 1<sup>st</sup> Law Efficiency vs. CR

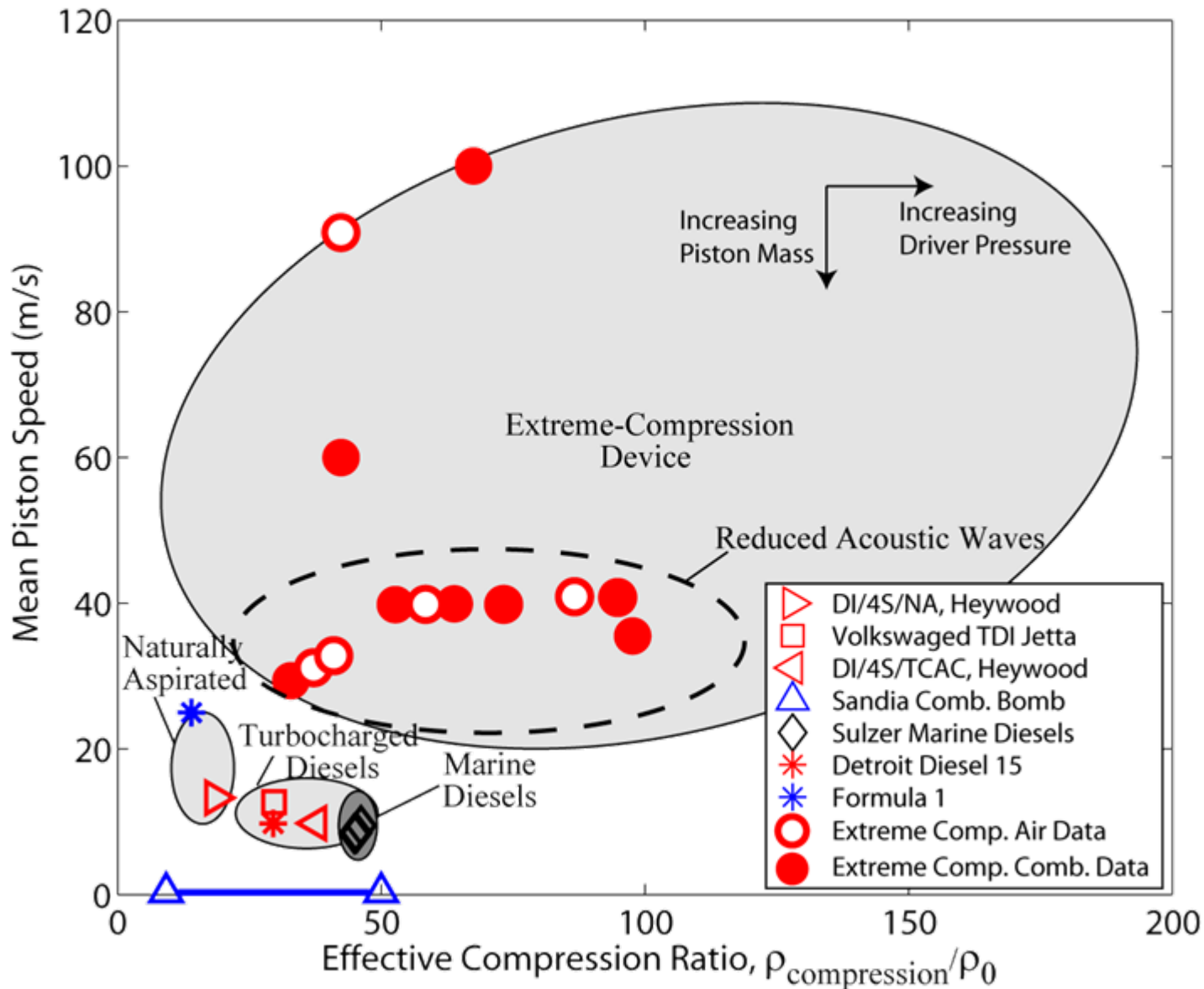


# Critical Questions Revisited

Critical Question	Extreme Compression Apparatus	Extreme Compression Engine
Material stress	✓	✓
Wall temperatures and heat transfer	■	■
NOx	■	■
Seal survivability	✓	■
Sealing ability	■	■
Ignition phasing	✓	✓
Combustion control	●	●

✓	solutions available and understood
■	more research required, but no obvious barriers
●	high priority for research

# Operating Space



# Next Steps

- Assess Combustion Performance
  - Injection performance at high densities
  - Ignition delay timing
  - Ability to generate turbulent mixing
  - Combustion efficiency
  - Implications for heat loss
- Install window to visualize spray dynamics and combustion
- Analysis and experimental testing of ring design to quantify and reduce blowby
- Install exhaust gas analysis system for measuring NO<sub>x</sub> and combustion efficiency

# Posters

- *Increasing Engine Efficiency through Extreme Compression:* S.L. Miller, M.N. Svrcek, K.-Y. Teh, O. Lacroix, J. Wilson, C.F. Edwards
- *Optimal Architecture for Efficient Steady-Flow Engines:* S. Ramakrishnan, K.-Y. Teh, S.L. Miller, C.F. Edwards